

This document was created on July 20th 2021 as a means of assurance that should the LOGS for Aircraft C-FZNR inadvertently become lost, that they can be recreated .

This digital copy includes the following: The Journey Log, The Engine Log, The propeller Log, Mogas, STC, Weight & Balance, record of Engine overhaul, and other miscellaneous documents found in the Document Pouch.

Logs were provided by Kelly Copp and digitized by Norman Fillion

CP-2NR



CANADA

AIRCRAFT
TECHNICAL LOG

DEPARTMENT OF TRANSPORT
CIVIL AVIATION BRANCH

MIDWEST AVIONICS

10-20 HANGAR LINE ROAD, WINNIPEG, MB R3J 3Y8

Work Order: 39251

Nomenclature: ELT

Manufacturer: AMERI-KING

Type/Model: AK-450

Part No:

Serial No: 485146

Inspected ☐ Function Test ☐

Modified ☐ Operational Test ☐

Repair ☐ Other Test ☐

Recertified ☒ Removal ☐

Previous Certification: INSPECTION DUE:

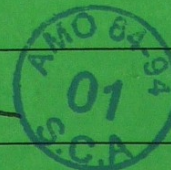
NOV 2018 BATTERY DUE: DEC 2024

The Maintenance described has been performed in accordance with applicable airworthiness requirements.

Date: 08 NOVEMBER 2017

AMO 84-94:

SERVICEABLE



Work Order

Aircraft: Piper Pa-22 Tri-Pacer
Reg: C-FZNR

July 05, 1999

Fuselage, Tail surfaces, Ailerons & Flaps

Rejuvenated old Poly-Tone and finished with Poly-Spray and Aero-Thane as per Stits Procedure Manual.
Replaced all windows.

Wings

Removed old fabric and inspected frame for wear and corrosion.
Replaced one section of Leading Edge and wing tip bows.
Removed dents and primed steel parts.
Covered wings with Poly-Fiber and finished with Aero-Thane as per Stits Procedure Manual.

Wheels

Replaced one tire and three inner tubes.

Wing Struts

Inspected and sealed by

Interior Pacific Flight Systems Ltd
STC SA4635NM

Engine

Overhauled by Universal Aero Engine.

Propeller

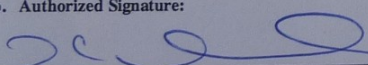
Overhauled by Maxwell.

Radios

Com, Transponder, Intercom, Loran C installed by G. Kunderman.

CR Blais

21061591

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: H-R101740	
4. Organization Name and Address: HARTZELL ENGINE TECHNOLOGIES, 2900 SELMA HWY, MONTGOMERY, AL. 36108					5. Work Order/Contract/Invoice Number: M162490	
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	ALTERNATOR-NEW **** END ****	SAL12-70	1	H-R101740	NEW	
12. Remarks: AIRWORTHINESS APPROVAL- PARTS.FOR PMA ELIGIBILITY SEE WWW.HARTZELL.AERO/PMA/						
13a. Certifies the items identified above were manufactured in conformity to: <input checked="" type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.				14a. <input type="checkbox"/> 14 CFR 43.9 Return to Service... <input type="checkbox"/> Other regulation specified in Block 12... Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature: 		13c. Approval/Authorization No.: 749188261		14b. Authorized Signature:		14c. Approval/Certificate No.:
13d. Name (Typed or Printed): MICHAEL C. STRICKLAND		13e. Date (dd/mm/yyyy): 23/OCT/2017		14d. Name (Typed or Printed):		14e. Date (dd/mm/yyyy):

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

Part No. 10-5001

Installation Instructions

1. Disconnect aircraft battery.
2. Install alternator per included drawing.
3. Refer to appropriate engine and airframe service manuals for belt tension and bolt torques.
4. Install battery wire with MS25171-2S terminal nipple on 6mm output terminal and torque to 50 in. lb.
5. Install ground wire to any of the three 5mm studs on rear of alternator and torque to 35 in. lb.
6. Install field wire with MS25171-1S terminal nipple to F1 terminal on rear of alternator and torque to 20 in. lb.
7. NOTE: F2 terminal to remain grounded with ground strap UNLESS aircraft voltage regulator is a type "A" regulator using a 2-wire field circuit, in this case remove and discard ground strap from F2 terminal and connect wiring from voltage regulator to F1 and F2 terminals, torque to 20 in. lb.
8. If aircraft is equipped with an "alternator out light" circuit, connect that wire to the AUX terminal and torque to 20 in. lbs. Other wise leave AUX terminal open.
9. Reconnect aircraft battery.
10. Start aircraft and check alternator output for proper operation.

Maintenance Instructions

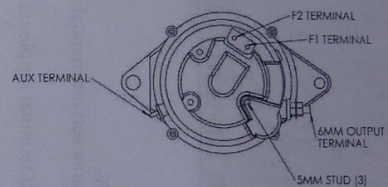
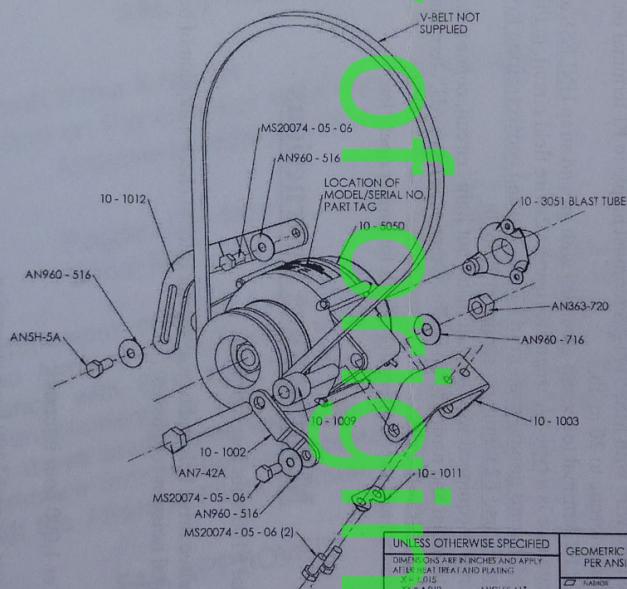
Annual / 100 hour inspections:

1. Remove drive belt and turn alternator rotor to check condition of bearings for abnormal noise or roughness.

5 year or 1,000 hour intervals:

1. Repeat: Annual / 100 hour inspection.
2. Remove field brush assembly and inspect brushes for excess wear. Replace brush assembly if brushes extend less than .250" from edge of brush holder case.

		REVISIONS		
EN	REV.	DESCRIPTION	BY	DATE
1409007	A	FIRST RELEASE INTO HET DESIGN DATA	B	9/3/74
1409008	B	1) THE BLOCK WAS PLANE POWER, TO 2) HARTZELL ENGINE TEC-HNOLOGIC PLANE POWER, TO 3) SHEET 2 WAS REVERSE SIDE 4) ADDED 10-103) CALL OUT TO DATA TAG 5) 0.5020 WAS ALG 704 REMOVED 6) DATA MATERIAL NOT FOR SHEET 1 - MAINTENANCE WAS OBTAINED AIRWORKINGNESS 7) ON SHEET 2 - MAINTENANCE INSTRUCTIONS WAS PLACED UNDER FOR CONTINUED AIRWORKINGNESS	B	9/3/74
1411045	C	ON SHEET 2 - MAINTENANCE INSTRUCTIONS WAS PLACED UNDER FOR CONTINUED AIRWORKINGNESS	OKO	11/19/74



SPECIFICATION CLASSIFICATION














CLASSIFICATION	DIMENSION CONVENTION	NOTE NO. CONVENTION
CRITICAL	<XXX.X>	[#]
MAJOR	[X.X.X]	[#]
MINOR	XXX.X	[#]
REFERENCE	(XX.X)	[#]

UNLESS OTHERWISE SPECIFIED

DIMENSIONS ARE IN INCHES AND APPLY
AFTER HEAT TREAT AND PLATING
X = $\pm .015$
XX = $\pm .010$ ANGLES $\pm 1^\circ$
XXX = $\pm .005$
BREAK ALL EDGES AND MACHINE ALL
INNER CORNER FILLETS .015 MAX.
SURFACE FINISH

THIS DRAWING CONTAINS INFORMATION THAT IS CONFIDENTIAL AND PROPRIETARY TO HARTZELL ENGINE TECHNOLOGIES. THE DRAWING IS RUSHED ON THE UNDERSTANDING THAT THE DRAWING AND THE INFORMATION IT CONTAINS WILL NOT BE COPIED OR DISCLOSED TO OTHERS EXCEPT WITH THE WRITTEN CONSENT OF HARTZELL ENGINE TECHNOLOGIES. WE WILL NOT BE USED TO THE DETRIMENT OF HARTZELL ENGINE TECHNOLOGIES, AND WILL BE RETURNED UPON REQUEST BY HARTZELL ENGINE TECHNOLOGIES.

GEOMETRIC SYMBOLS
PER ANSI Y14.5

	FLATNESS
	SIRKAGHNESS
	ROUNDNESS
	CYLINDRICITY
	PROMISE
	PERPENDICULARITY
	POSISH
	CONCENTRICITY
	SYMMETRY
	ANGULARITY
	PARALLELISM
	CIRCULAR RUNOUT
	TOTAL RUNOUT

SCALE	NTS
-------	-----

DRAWN	BJ	12/26/05
-------	----	----------

CHECKED	EAB	11/19/14
---------	-----	----------

ENG.	RFD	11/19/14
------	-----	----------

FINISH	N/A
--------	-----

WEIGHT	N/A lbs
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MATERIAL
SEE INDIVIDUAL

SEE INDIVIDUAL COMPONENTS

SIZE	SH 1	CODE ID
B	Q50	15BY1

B	OF 2	65PYI
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HARTZELL
ENGINE TECHNOLOGIES

2900 Selma Highway
Montgomery, AL 36108

AL24-70 INSTALLATION INSTRUCTIONS

DRAWING NO.

	R
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REV.

INSTALLATION AND MAINTENANCE
INSTRUCTIONS ON SHEET 2



Transport Canada
Transports Canada

ANNUAL AIRWORTHINESS INFORMATION REPORT

COPP, KELLY L

92 ELM DRIVE

OAKBANK, MANITOBA, CANADA

PNR 1.72

COMPLETE FORM AND SUBMIT NO LATER THAN THE DUE DATE

See reverse for change of address and form instructions.

Aircraft operated pursuant to CAR IV or CAR VII (inspection information not required.)	
Date of the most recent annual or 100 hour inspection (yyyy-mm-dd) 2019-10-25	
AMO, AME or owner who/which conducted & certified this inspection	
Name	<input type="radio"/> AMO Number
COPP, KELLY L	<input type="radio"/> AME
	<input checked="" type="radio"/> Owner
Has the aircraft been damaged since last report? Date of damage (yyyy-mm-dd)	
<input type="radio"/> Yes (if yes, enter date)	<input checked="" type="radio"/> No
Date of repair certification (yyyy-mm-dd):	
Owner's Contact Information	
Fax Number (999-999-9999): 204-444-3283	
E-Mail Address: KELLYCOPP@MYMTS.NET	

Issued (yyyy-mm-dd) 2020-09-02	Due (yyyy-mm-dd) 2020-03-30
Registration Mark FZNR	Type Certificate number NOT CERTIFICATED
Type of Flight Authority OWNER MAINT	Type of Registration PRIVATE

AIRCRAFT BASE OF OPERATION	
Country CANADA	Province/State MANITOBA
Municipality OAKBANK	Airport ST. ANDREWS
Other (House/Farm/etc) 24	TC Region PNR

Reporting year 2019	
Total hours flown since new – to Dec 31 of the reporting year 3565	Hours flown – Jan 01 to Dec 31 of the reporting year 0
Optional for air operators & flight training units	
Specialty Hours Training 0	Other aerial work 0

Aircraft Make PIPER	Aircraft Model PA22 150 X	Aircraft Serial Number 223388	
Empty Weight Last Actual 1164		Last Amended 0	
Which landing gear configuration? Date (yyyy-mm-dd) 1b <input checked="" type="checkbox"/> Wheels <input type="checkbox"/> Floats <input type="checkbox"/> Skis 2008-06-12		Which landing gear configuration? Date (yyyy-mm-dd) 1b <input checked="" type="checkbox"/> Wheels <input type="checkbox"/> Floats <input type="checkbox"/> Skis	
Max. permissible take-off weight 2000		Which landing gear configuration? 1b <input checked="" type="checkbox"/> Wheels <input type="checkbox"/> Floats <input type="checkbox"/> Skis	
Engine Make AVCO LYCOMING		Propeller Make	
Engine Model 1. O-320 X	Engine Serial Number 1. L576527	Propeller Model 1.	Propeller Serial Number 1.
Engine Model 2.	Engine Serial Number 2.	Propeller Model 2.	Propeller Serial Number 2.
Engine Model 3.	Engine Serial Number 3.	Propeller Model 3.	Propeller Serial Number 3.
Engine Model 4.	Engine Serial Number 4.	Propeller Model 4.	Propeller Serial Number 4.
Ski Make	Ski Model	Float Make	Float Model

Owner Remarks

AIRCRAFT OUT OF SERVICE <input type="checkbox"/>
Note: Hours flown and TSN must be completed up to the date of out of service.
This aircraft will be/has been out of service from (yyyy-mm-dd) _____
Estimated date of return to service (yyyy-mm-dd) _____
Reason (optional) _____

I CERTIFY THAT THE INFORMATION SUPPLIED IS CORRECT

Name of owner or authorized agent (please print) _____ Date (yyyy-mm-dd) _____

Signature of owner or authorized agent _____

Transport Canada (AARDA)
330 Sparks Street
Place de Ville, Tower C
Ottawa, Ontario K1A 0N8

ANNUAL AIRWORTHINESS INFORMATION REPORT (AAIR)

Subject to Canadian Aviation Regulations (CARs), Part V, Subpart 1, and Chapter 501 of the Airworthiness Manual, the owner of a Canadian aircraft, other than a light aircraft, shall submit to the Minister no later than the due date an Annual Airworthiness Information Report (AAIR).
For any questions related to this form, please contact Transport Canada Headquarters at 1-888-663-3639 (Option 1) or by email at tc.aair-naia.tc@tc.gc.ca

AIRCRAFT REGISTRATION

For all changes to your Aircraft Registration including but not limited to: change of address, aircraft sold, aircraft destroyed, aircraft retired, etc., contact your TC Regional Office at 1-800-305-2059, option 1 for English, 2 for Aircraft Registration and then select the appropriate region. **Indicating this type of information on your AAIR form (hard copy or online) will not update your Aircraft Registration.**

HOW TO SUBMIT THE AAIR

The AAIR can be submitted in one of two ways:

1. Fill in the attached form and return it to Transport Canada (TC), following the instructions of sections A, B and C below; or
2. Submit it electronically via the Internet, following the instructions of section D below.

A. PROVIDE DATA IN THE BLANK SPACES

1. Date of the most recent annual or 100-hour inspection, and the name & number of the AMO or AME who/which conducted and certified the inspection. In the case of an amateur-built or owner maintained aircraft, the name of the owner if the owner performed the inspection.

Note: Not required for aircraft operated pursuant to CAR IV or CAR VII.

Note: The AAIR due date and the aircraft's inspection date are not related.

2. Indicate (yes or no) whether the aircraft was significantly damaged since the last report. If YES, the date the aircraft was damaged and, if applicable, the date of the repair certification.

3. Aircraft total hours flown since new – to December 31st of the reporting year, and aircraft hours flown from January 1st to December 31st of the reporting year.

Note: Air operators and flight training units may provide a consolidated breakdown of the total training hours and the total hours flown for other aerial work activities (specialized work) in the reporting year.

4. Provide the aircraft owner's 24-hour fax number and/or e-mail address to which TC can forward notifications of applicable Airworthiness Directives (ADs) and Civil Aviation Safety Alerts (CASAs), as well as to address matters related to AAIRs.

B. AIRCRAFT OUT OF SERVICE

Subject to certain conditions, an AAIR is not required for an aircraft that is out of service (not flown) for all of a complete calendar year. When applicable, an owner can claim this privilege by indicating on this form:

1. Check the box in the AIRCRAFT OUT OF SERVICE section; and
2. Indicate the date the aircraft will be/has been out of service from; and
3. Indicate the estimated date of return to service. If unknown, indicate an approximate date.

Note: If the aircraft is declared out of service part way into a calendar year, an AAIR will be sent the following year in order to capture the hours flown, and other related data, from January 1st up to the date the aircraft was declared out of service.

C. MAILING INSTRUCTIONS

1. Check that all necessary data has been supplied.
2. Print the form, sign and date the lower right-hand corner.
3. Mail the form to:
Transport Canada (AARDA)
330 Sparks St, Place de Ville, Tower C
Ottawa, Ontario K1A 0N5
4. Add sufficient postage and ensure that the envelope is postmarked no later than the due date.

D. INTERNET SUBMISSION INSTRUCTIONS

1. Using your AAIR User ID, log in to the Continuing Airworthiness Web Information System (CAWIS) at www.tc.gc.ca/cawis-swimn (Use of your AAIR User ID is deemed to be the equivalent of your signature when submitting an AAIR via the Internet.)

Note: User IDs are assigned by TC and will remain the same each year. If you do not have a User ID or have forgotten it, you can retrieve your User ID and create a password by selecting the "Forgot User ID or Password" link on the AAIR Login Page.

2. Check and update all existing data, and enter the required new data following the guidance of sections A and B above.
3. Click the "Submit" button at the bottom of the page to send your completed AAIR to Transport Canada. The page will refresh, confirming your AAIR was successfully submitted.
4. Click the "Print" button at the bottom of the page if you would like to keep a hardcopy for your records.

Transport Canada Transports Canada

Thank You.

The AAIR has been successfully submitted:

2020-09-02

Date (yyyy-mm-dd)

If there are any discrepancies with the information you have provided, Transport Canada will be in contact with you. It is recommended that you now print or save this acknowledgement as proof of compliance with CAR 501.

VOLUME NO.

SECTION II

AIRFRAME

LOG

PRICE \$1.00

AIRCRAFT

1. Nationality and Registration

~~C-FZNR~~
~~CF-ZNR~~

2. Manufacturer's Designation

PA-23

3. Type Approval or
Specification Number

1A6

4. Manufacturer's Serial Number

22-3388

5. Date of Manufacture

1955

EMPTY WT - 1142[#]
MAX GROSS - 2000[#]

ENGINEER

[illegible]

SECTION II

RECORD OF INSTALLATIONS
AND MODIFICATIONS
TO AIRCRAFT

[illegible]

A collage of various vehicle inspection and repair forms. The top part shows a 'VEHICLE INSPECTION REPORT' with fields for 'VEHICLE IDENTIFICATION' and 'DATE OF INSPECTION'. Below that is a 'VEHICLE REPAIR ORDER' form with fields for 'VEHICLE IDENTIFICATION', 'DATE OF ORDER', and 'VEHICLE REPAIR ORDER'. The bottom part shows a 'VEHICLE INSPECTION REPORT' with fields for 'VEHICLE IDENTIFICATION', 'DATE OF INSPECTION', and 'VEHICLE REPAIR ORDER'. The forms are overlaid with a large, semi-transparent watermark that reads 'Copy of original documents / info'.

A collage of various vehicle inspection and repair forms. The top part shows a 'VEHICLE INSPECTION REPORT' with fields for 'VEHICLE IDENTIFICATION' and 'DATE OF INSPECTION'. Below this is a 'VEHICLE REPAIR ORDER' form with fields for 'VEHICLE IDENTIFICATION', 'DATE OF ORDER', and 'REPAIRS'. The bottom part shows a 'VEHICLE INSPECTION REPORT' with fields for 'VEHICLE IDENTIFICATION', 'DATE OF INSPECTION', and 'REPAIRS'. The forms are overlaid with a large, semi-transparent watermark that reads 'Copy of original documents / info'.

ENGINE &
POSITION DATE
INSTALLED
19615
0325FA
No. 19098
17412M
Repaired
Inspector's Signature, Stamp
19098
inspected
e Federal
to service.
ncy under
26-92
FOR
C.
55429

[illegible]

PROPELLER LOG

1. Make SENSENICH
2. Model M74 DM 58.
3. Specification No. _____
4. Date of Manufacture _____
5. Hub Serial No. 19098
6. Blades
 - (a) Design No. _____
 - (b) Blade Serial Nos.
 - (i) _____
 - (ii) _____
 - (iii) _____
 - (iv) _____
 - (c) Pitch Setting
 - (i) Basic _____
 - (ii) High _____
 - (iii) Low _____

PROP. BOLT TORQUE 23-25 FT.-LBS.

SECTION III

ENGINE

LOG

ENGINE LOG

1. Make
2. Model
3. Specification
4. Mfg. Serial No.
5. Reduction Gear Ratio
6. Date of Manufacture

LYCOMING

O-320

L-5765-27

DATE	HOURS	MIN	SEC	REMARKS
27 July 87				LIH MAG LEAD REPAIRED. GENERATOR REPAIRED. REGULATOR PUMPS CLEANED. PROPS SEAL REPLACED (C/I) REPAIR.
5 Aug 88	5	3	552	2 CCI
4 Aug 89	2	0	554	2 CCI
MAY 30-90	8	8	5630	
APR 7-91	41.2		604.2	
Aug 10/91			612.4	OIL CHANGE aircraft using Mogas
Aug 10/91			612.4	100 commercial inspection
NOV 28-91	13.3		615.7	OIL CHANGE PHILIPS 20/50

TOTAL THIS PAGE

CARRIED FORWARD

ENTRIES DRAWN AT

SCHEMATIC IS AIRWORTHY

I hereby certify I have completed the nearest eq to a 24 hr. periodic inspection of the aircraft described herein and it is airworthy.

I certify I have inspected this aircraft in compliance with the Condition and Conformity Inspection procedure described in the Engineering and Inspection Manual and it is airworthy.

I certify I have inspected this aircraft in compliance with the Condition and Conformity Inspection procedure described in the Engineering and Inspection Manual and it is airworthy.

Replac Phg Harness
Compressor Test 1 74 2 69 3 78
Oil Brown 1. Chng Oil Shell WIC

under STC no 5A8496L
aircraft certified airworthy

July 10/99	639.9	engine reinstalled	UNIVERSAL AERO ENGINES LTD. WINNIPEG, ST. ANDREWS, MANITOBA DOT APPROVED F UAE	
JULY 21-00	718.2	CHANGED TO OWNER MAINTENANCE	engine AD new bolts certified air see work CATEGORY	
AUG. 4-00	718.2	ANNUAL INSPECTION		
		#1-80/74 #2-80/75 #3-80/73 #4-80/77		

MAINTENANCE RECORD		SIGNATURE	LICENCE NUMBER
REPAIRS, ADJUSTMENTS, MODIFICATIONS ENTRIES. DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.			
ENGINE [REDACTED] / REPAIRED AND TESTED. IN ACCORDANCE WITH MANUFACTURERS SPECIFIC- INSTRUCTIONS COPIES OF MOD'S AND PARTS LIST ATT- ACHED. CERTIFIED AIRWORTHY			
DATE <u>MAY 3 1999</u> UNIVERSAL AERO ENG. LTD. F UAE <u>871</u> <u>81-91</u>		<i>[Signature]</i>	M37675U
[REDACTED] completed by Universal engine AD 98-02-08 C/W new bolts installed & torqued certified annually see work order CATEGORY		<i>[Signature]</i>	AR Blair M067591
		<i>[Signature]</i>	PA 36082

MANITOBA HYDRO

FROM OUR OFFICE AT

C - F Z N R

Recovered with-

Dacron Fabric - Style D - 101 A 3.7 oz. per. yd.

Finished with -

Stits Poly Fibre Aircraft Covering Process,

Poly -brush and Poly - spray

Painted with -

~~Stits Poly~~

Stits Poly-tone, Wings and fuselage Juneau White # 110

Trim - Tennessee Red # 190

Miami Blue # 175

Registration changed from CF-ZNR to C-FZNR

E L T installed in baggage compartment.

Copy of original documents / nt

PIPER AIRCRAFT CORPORATION INSPECTION REPORT THIS FORM MEETS REQUIREMENTS OF FAR PART 43					
Make PIPER TRI-PACER AND CARIBBEAN		Model PA-22-125, PA-22-135 PA-22-150, PA-22-160		Serial No. 22-3208 X	Registration No. FZNR
Circle Type of Inspection (SEE NOTE 1, PAGE 24) 50 100 500 1000 Annual		Inspector		Perform inspection or operation at each of the inspection intervals as indicated by a circle <input type="radio"/>	
DESCRIPTION		50	100	500	1000
A. PROPELLER GROUP					
1. Inspect spinner and black plate		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
2. Inspect blades for nicks and cracks		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
3. Check for grease and oil leaks		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
4. Lubricate propeller per lubrication chart		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
5. Check spinner mounting brackets		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
6. Check propeller mounting bolts and safety (Check torque if safety is broken.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
7. Inspect hub parts for cracks and corrosion		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
8. Rotate blades of constant speed propeller and check for tightness in hub pilot tube		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
9. Remove constant speed propeller, remove sludge from propeller and crankshaft		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
10. Overhaul propeller		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
B. ENGINE GROUP					
CAUTION: Ground Magneto Primary Circuit before working on engine.					
1. Remove engine cowl		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
2. Clean and check cowling for cracks, distortion and loose or missing fasteners		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
3. Drain oil sump		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
4. Clean suction oil strainer at oil change (Check strainer for foreign particles.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
5. Clean pressure oil strainer (Check strained for foreign particles.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
6. Check oil temperature sender unit for leaks and security		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
7. Check oil lines and fittings for leaks, security, chafing, dents and cracks.		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
8. Clean and check oil radiator cooling fins for damage		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
9. Remove and flush oil radiator		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
10. Fill engine with oil per lubrication chart		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
11. Clean engine		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
12. Check condition of spark plugs (Clean and adjust gap as required, adjust per Lycoming Service Instruction No. 1042.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
13. Check ignition harness and insulators (High tension leakage and continuity.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
14. Check magneto points for proper clearance - Maintain clearance at 0.018 ± 0.006		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
15. Check magneto for oil seal leakage		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
16. Check breaker felts for proper lubrication		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
17. Check distributor block for cracks, burned areas or corrosion, and height of contact springs		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
18. Check magnetos to engine timing		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
19. Overhaul or replace magnetos (SEE NOTE 2, PAGE 3)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
20. Check valve clearance at 0.010 on O-290-D engine only (Adjust per Lycoming Service Instruction No. 1068)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
21. Remove air filter and clean (Refer to Owner's Handbook.) (Replace as required.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
22. Drain carburetor and clean inlet line fuel strainer		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
23. Check condition of carburetor heat air door and box		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
24. Check intake seals for leaks and clamps for tightness		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
25. Remove, drain and clean fuel filter bowl and screen (Drain and clean at least every 90 days.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
26. Check condition of flexible fuel and primer lines		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
27. Replace flexible fuel lines (SEE NOTE 2)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
28. Check fuel system for leaks		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
29. Check venturi or vacuum pump, lines and separator		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
30. Overhaul or replace vacuum pump (SEE NOTE 2)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
31. Check throttle, carburetor heat, and mixture controls for travel and operating condition		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
32. Inspect exhaust stacks, connections and gaskets (Replace exhaust gaskets as required.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
33. Inspect muffler, heat exchanger and baffles (Refer to Piper Service Letter No. 324B.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
34. Check exhaust stack braces.		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
35. Check breather tube for obstructions and security		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
36. Check crankcase for cracks, leaks and security of seam bolts		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
37. Check engine mounts for cracks and loose mountings		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
38. Check all engine baffles for damage and security		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
39. Check rubber engine mount bushings for deterioration (Refer to Piper Service Letter No. 223.) (See Note 3.)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
40. Check condition of firewall seals		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
41. Check condition and tension of generator drive belt		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
42. Check condition of generator and starter		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
43. Lubricate all controls		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
44. Complete overhaul of engine or replace with factory rebuilt (SEE NOTE 2)		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>
45. Reinstall engine cowl		<input type="radio"/>	<input type="radio"/>	<input type="radio"/>	<input type="radio"/>

Owner: _____

Continued on reverse side

230 201 710817

Circle Type of Inspection (SEE NOTE 1, PAGE 3)		Inspector				Perform inspection or operation at each of the inspection intervals as indicated by a circle (○).			
50	100	500	1000	Annual	50	100	500	1000	
C. CABIN GROUP					E. WING GROUP				
1. Inspect cabin entrance, doors, latches and windows for damage and operation. 2. Check all plexiglas for cracks 3. Check upholstery for tears 4. Check seats, seat belts, security brackets and bolts 5. Check trim operation and adjustment 6. Check rudder pedals 7. Check control yoke, chain, pulleys and cables 8. Check flap lever for operation, adjustment and safety 9. Check controls for ease of operation 10. Check battery, box and cables (Check at least every 30 days. Flush box as required and fill battery per instructions on box.) 11. Check landing, navigation, cabin and instrument lights 12. Check fuse box for burned out fuses 13. Check instruments, lines and attachments 14. Check gyro operated instruments and electric turn and bank (Overhaul or replace as required.) 15. Replace filters on gyro horizon and directional gyro or replace central air filter 16. Clean or replace vacuum regulator filter 17. Check altimeter (Calibrate altimeter system in accordance with FAR 91.170, if appropriate.) 18. Check operation of fuel selector valve (Refer to Piper Service Letter No. 351.) 19. Remove, drain and clean right fuel tank filter bowl and screen (Drain and clean at least every 90 days.) 20. Check condition of heater control and duct 21. Check condition and operation of air vents					13. Check stabilizer yoke and screw for end play and security 14. Check stabilizer attachments and attachment tube for side play. 15. Check stabilizer brace wires for corrosion, tightness and safety 16. Check elevator attachment and horn for damage 17. Check elevator hinge pins and bushings for excess wear and corrosion (Replace pins and/or bushings as required.) 18. Lubricate per lubrication chart 19. Reinstall inspection plates and panels				
D. FUSELAGE AND EMPENNAGE GROUP					F. LANDING GEAR GROUP				
1. Remove inspection plates and panels 2. Check fabric and finish for cracks and deterioration (If condition of fabric is doubtful, refer to CAM 18 or FAA AC 43.13-1. Use strip test method.) 3. Inspect fuselage fabric in area of windshield top attachment channel (Refer to Piper Service Bulletin No. 174.) 4. Check electronic installations for security 5. Check antenna mounts and electric wiring for damaged insulation and security 6. Check rotating beacon for security and operation 7. Check fuel lines for security and damage 8. Check rudder, elevator and stabilizer trim cables, turnbuckles, guides and pulleys for safety, damage, corrosion and operation 9. Check fuselage longerons and stringers for damage 10. Check rudder, stabilizer and elevator structures for damage 11. Check rudder attachments and horn for damage 12. Check rudder hinge pins and bushings for excess wear and corrosion (Replace pins and/or bushings as required.)					1. Remove fairings 2. Check fabric and finish for cracks and deterioration 3. Inspect gear and shock strut bolts and nuts for safety 4. Hoist airplane, check gear and shock strut bolts and bushings for excess wear and corrosion (Replace bolts and/or bushings as required.) 5. Inspect shock cords for broken threads and weakness, and shock struts for weakness (Replace cords and/or shock struts as necessary.) 6. Check main wheel alignment (0° Toe in - Toe out)				

Circle Type of Inspection (SEE NOTE 1, PAGE 3)		Perform inspection or operation at each of the inspection intervals as indicated by a circle (○).				
50	100	500	1000	Annual		
F. LANDING GEAR GROUP (cont)						
7. Check nose gear alignment, steering control and travel	○	○	○	○	○	
8. Check shimmy dampener for alignment and operation	○	○	○	○	○	
9. Check nose gear oleo strut for proper extension (3.5 in.) (Check for proper fluid level as required.)	○	○	○	○	○	
10. Check nose gear oleo strut for fluid leaks and scoring.	○	○	○	○	○	
11. Check nose gear struts, attachments, torque links, and bolts and bushings for condition and security	○	○	○	○	○	
12. Replace torque link and steering horn bolts and bushings	○	○	○	○	○	
13. Check tires for cuts, uneven or excessive wear and slippage.	○	○	○	○	○	
14. Remove wheels, clean, check and repack bearings	○	○	○	○	○	
15. Check wheels for cracks, corrosion and broken bolts	○	○	○	○	○	
16. Check tire pressure (Nose-15 psi / Main-22 psi)	○	○	○	○	○	
17. Check brake lining and disc for excessive wear	○	○	○	○	○	
18. Check brake lines for chafing and security	○	○	○	○	○	
19. Check brake cylinders, and parking valve for operation and leaks (Check fluid level as required.)	○	○	○	○	○	
20. Lubricate per lubrication chart.	○	○	○	○	○	
21. Reinstall fairings.	○	○	○	○	○	
G. FLOAT GROUP						
1. Check float attachment fittings	○	○	○	○	○	
2. Check floats for damage	○	○	○	○	○	
3. Check pulleys and cables	○	○	○	○	○	
H. OPERATIONAL INSPECTION						
1. Check fuel tank selector	○	○	○	○	○	
2. Check fuel quantity	○	○	○	○	○	
3. Check oil pressure and temperature	○	○	○	○	○	
4. Check generator output	○	○	○	○	○	
5. Check carburetor heat	○	○	○	○	○	
6. Check parking brake	○	○	○	○	○	
7. Check vacuum gauge	○	○	○	○	○	
8. Check gyros for noise and roughness	○	○	○	○	○	
9. Check cabin heater operation	○	○	○	○	○	
10. Check magneto switch operation.	○	○	○	○	○	
11. Check magneto RPM variation	○	○	○	○	○	
12. Check throttle and mixture operation	○	○	○	○	○	
13. Check propeller smoothness.	○	○	○	○	○	
14. Check propeller governor action (constant speed)	○	○	○	○	○	
15. Check electronic equipment operation	○	○	○	○	○	
16. Check engine idle	○	○	○	○	○	
I. GENERAL						
1. Aircraft conforms to FAA Specifications	○	○	○	○	○	
2. All FAA Airworthiness Directives complied with	○	○	○	○	○	
3. All Manufacturers Service Letters and Bulletins complied with	○	○	○	○	○	
4. Check for proper Flight Manual.	○	○	○	○	○	
5. Aircraft papers in proper order.	○	○	○	○	○	
NOTES: 1. Both the annual and 100 hour inspections are complete inspections of the airplane - identical in scope. Inspections must be accomplished by persons authorized by FAA. 2. Replace or overhaul as required or at engine overhaul. (For engine overhaul, refer to Lycoming Service Instructions No. 1009.) 3. It is recommended that all engine mount rubber bushings be replaced every five hundred hours.						
REMARKS:						
Signature of Mechanic or Inspector <i>W. J. Smith</i>		Certificate No.		Date <i>AUG 4-00</i>		Total Time on Airplane <i>2916.1</i>

Canadian Tire Store #447
1041 Manitoba Avenue
Selkirk MB R2G 1A 3T7
Phone: 204-482-4400 Fax: 204-482-4726
E-mail: autoservice447@ctis.net

INVOICE

10044700018667
(Copy)

GST REG#: 866710286R

PST REG#: 044232-7

SOLD TO:
17904339.0447
W BIRCH
112-1100 HENDERSON HWY
WINNIPEG MB R2G 1L2

MOBILE :
BUSINESS:
EMAIL :

HOME: 204 561-1021
FAX :

COLOUR	VEHICLE DESCRIPTION	PLATE	UNIT#	TAG	ODO IN
					0
VIN	VEHICLE OPTIONS			ADV	ODO AUTH
				MO1	0
TIME IN	PROMISED	TERMS	GST EXEMPT#	P.O.	ODO OUT
15:38		Cash			0

QTY	ITEM	DESCRIPTION	WARR	NET	EXT PRICE
1	0102050	MOTOMASTER MM-UI L&G.BAT MM-UI L&G.BAT Battery S/N: 102050 Core Exchange 0102050 MOTOMASTER MM-UI L&G.BAT MM-UI L&G.BAT --serial#: 10044700018667-01	MILES #7 54.99 OPERATOR #: 0123 \$ 22 10/05/2015 14:41 REG # 7	54.99	54.99
-1	XCORE		--PARTS ORDER #:10044700018667-- 1X01020504 \$ 74.99 TRADE-IN \$ -20.00 --END OF ORDER #:10044700018667--		Included
		Thank you for having your vehicle serviced at Canadian Tire Selkirk. If your tire's were removed during your visit, please return for a curtesy wheel retorque within 100km or 3 day's.			
			SUBTOTAL \$ 54.99 GST 5% \$ 2.75 PST 8% \$ 4.40 TOTAL \$ 62.14 CTM earned on \$ 54.99 CASH TEND \$ 65.20 CASH TND RNDNG \$ -0.01 CHANGE \$ 3.05 BASE CT MONEY \$ 0.25		

*** Warranty Code Legend ***

M1: See Service Manager for details

*** Global Warranty Message ***

Warranties applicable only when product is installed at a Canadian Tire Automotive Service Centre, otherwise warranties may vary. Warranty not available for commercial use. A minimum labour warranty of 100 days / 5500 km applies to parts installed unless otherwise stated above. Parts Warranty starts on the completion date shown on the customer invoice.

PRIVACY TERMS

Canadian Tire Corporation, Limited, its affiliates, Associate Dealers, and service providers ("CTC" or "our") collect and use the personal information provided by you to fulfill this contract, CTC may also collect and use your information to administer the delivery of and market warranties, products, services, and loyalty programs to you, and to analyze your relationship with us. Please visit www.canadiantire.ca for details on CTC's privacy policy. If you do not wish for your information to be kept by CTC, please contact CTC's Corporate Customer Relations at 1-866-846-8468 or <https://www.ctis.com/ContactUs/>.

Shop supplies are calculated at a rate of 10% of labour and parts. This means that you have rights and responsibilities under the stated, estimated protection and you have rights and responsibilities under the work and repairs. A written estimate for repairs that cost more than \$100 and exceeding a written estimate, and you specifically authorizing your repair and the cost charged for the work or repairs did not exceed your authorized repair estimate unless you were told about the fee and agreed to pay it. You cannot

Product Number / Numéro d'article
010-2050-4
Serial Number / Numéro de série
01215490



INVOICE TO:

I hereby authorize you and your employees to carry out the repairwork described above and to purchase on my account any parts and materials necessary to carry out the repairwork subject to the estimate described above. I hereby authorize you and your employees to operate the vehicle listed above in connection with the repairwork, including operation for the purposes of testing, inspection or delivery. I HEREBY RELEASE AND FOREVER DISCHARGE YOU AND EACH OF YOUR EMPLOYEES FROM ANY LOSSES I MAY SUFFER RELATING TO DAMAGE TO OR LOSS OF MY VEHICLE AND/OR ANY ITEMS CONTAINED THEREIN THAT ARE CAUSED BY CIRCUMSTANCES BEYOND YOUR CONTROL. I acknowledge that, upon the completion of the repairwork authorized hereby, you will have a lien pursuant to applicable repair and storage lien legislation and that you may register that lien and seize, at your discretion and at my sole cost and expense, the vehicle for non-payment of any invoice issued for such repairwork. By signing below I acknowledge and agree to the terms and conditions above and to those printed on the reverse side of this form.

Customer Signature: _____

PARTS:	
LABOUR:	
OTHER:	
SUB-TOTAL:	
GST/HST:	
PST:	
TOTAL:	Cont'd

CANADIAN TIRE
SELKIRK #477

204-482-8473

PRICE ADJ MAY BE PERMITTED WITHIN 7 DAYS
SEE BACK OF RECEIPT FOR RETURN INFO.

GST # 868 710 286

REG #: 7 10/05/2015 14:41:22 TRANS #: 22

OPERATOR #: 0123 Float: 001

==PARTS ORDER #: 10044700018667==

010-2050-4 MOTOMASTER MM-U \$ 74.99

TRADE-IN \$ -20.00

==END OF ORDER #: 10044700018667==

SUBTOTAL	\$	54.99
GST 5%	\$	2.75
RST 8%	\$	4.40
T O T A L	\$	62.14
CASH TEND.	\$	65.20
CHANGE	\$	3.05
BASE CT MONEY	\$	0.25

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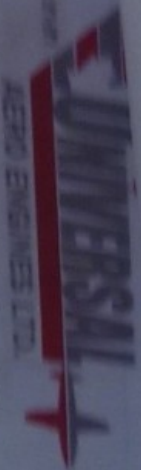
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to legal residents of Canada, age of
majority or older. Math skill-test
required. Odds of winning depend on #
of entries received. Conditions apply.
For contest rules visit website.

0002-7041-24500-11234



0044715100540123000000010022



11, Banger's Place, St. Andrews, Northdown, P.O. Box 391
Phone: 0204 735-4072 Fax: 0204 735-1228

Owner: Wally Birch
Date: September 17, 2010
Make: Lycoming
Model: O-320
Serial #: L-5765-27X

PARTS		DESCRIPTION	
QTY	PART#	QTY	DESCRIPTION
1	185M-0207-20	1	Crankshaft Kit (LM-1185M)
	Crankshaft Series # 174528		Cylinder #0 Series # A04Q1103
4	059C142204	4	Cylinder Assy. (185C14508)
	Cylinder #0 Series # A04Q1103		
1	100204-25	1	Accessory Case
2	10-471805	2	Bearing Ball Drive End
2	25-2012	2	Bearing Ball Top
3	18M130385-WC10	4	Bearing Con-Rod
2	18M26035-WC10	3	Bearing Front Main
4	18L026100-WC10	1	Bearing Main (Main Wall)
3	155250564	3	Body Hydraulic tappet
3	15L-78027	1	Bot. Con-Rod
1	5STD-2245	4	Bot. Crankshaft Gear
2	10-1650644	4	Brush Carbon
4	15L-78385	3	Exhaust Valve Rocker Arm
4	15L-78387	2	Exhaust Valve Rocker Shaft
3	15L-78386	2	Cap. Exhaust Rotator
4	17C13385	24	Capacitor
2	10-51673	2	Cart. Quick Kit
1	AV2285-04	2	Contact Point
2	10-357114	4	Data Plate Carburetor
1	WE0015	2	Data Plate Magnet
2	WE0014	3	Flower Kit Small WA
1	1855-1000-F	1	Gasket Set Major (OH)
1	19L-55371-1	2	Hose Intake Pipe
4	15L-5TD-1195	4	Hose Oil Drain Line
4	15L-5TD-1821	2	Nut 1/4-18
40	15L-5TD-1411	2	Nut 5/16-18
24	15L-5TD-1410		

Email: info@universalengines.com

Form# LUE-35 R010152010

copy of original documents /nf



10 Prairie Place, St. Andrews, Manitoba, R1A 3P1
Phone: (204) 339-4078; Fax: (204) 338-2230

ACCESSORY DATA WO#3287

Owner: Wally Birch

Date: September 17, 2010

Make: Lycoming

Model: 0-320

Serial #: L-5765-27X

Description	Make	Model / Part #	Serial #
Camshaft	Lycoming	LW-18840	7458
Carburetor	Marvel-Shebler	MA-4SPA / 10-36781-1	3980506
Crankcase	Lycoming	0-320 (ND)	None Found
Crankshaft	Lycoming	None Found	None Found
Cylinder #1	Lycoming	16B21508	A0401013
Cylinder #2	Lycoming	16B21508	A042108
Cylinder #3	Lycoming	16B21508	A078101
Cylinder #4	Lycoming	16B21508	A042103
Magneto Right	Bendix	S4LN-20 / 10-51360-29	3249
Magneto Left	Bendix	S4LN-21 / 10-51360-37	846167
Vacuum Pump	PESCO	3P194F	B7979M

E-mail: info@universal aeroengines.com

Form# UAE-26 R0 03/15/2010

Web-Site: www.universal aeroengines.com



AD Notes Compliance Record WO#3287

AD#	Applicable S.B.# & Subject	Date & Hours @ Comp	Method of Compliance	One time	Recurring	Next Comp. @ Hrs/Date	Authorized Signature & Number
55-02-02	Drive adaptor gasket		Embodied	X			UAE-2
59-10-07	Cylinder baffle clamps		Found Embodied	X			UAE-2
63-22-03	Marvel Schebler carburetors		Found Embodied	X			UAE-2
63-23-02	Exhaust valve stem		N/A Has 1/2 inch Exhaust Valve & N/A by Camshaft Part #		X		UAE-2
64-16-05	Oil seal failure		N/A No Fuel Pump Installed	X			UAE-2
66-20-04	Oil filter adapter gasket		N/A by Casper Labs Inc. STC#01930NY	X			UAE-2
69-24-03	Marvel Schebler carburetors		N/A by Carburetor Serial #	X			UAE-2
72-06-05 R2	Marvel Schebler carburetors		N/A by Throttle Arm Part #	X			UAE-2
73-07-04	Bendix Magnetos		Superseded by 94-01-03 R2				UAE-2
73-23-01	Piston pin failures		N/A by Piston Pin Part #	X			UAE-2
74-26-09	Bendix Magnetos		(L/H) Found Embodied-(R/H) N/A by Magneto Model #	X			UAE-2
75-08-09	Oil pump		Found Embodied	X			UAE-2
78-09-07 R3	Bendix Magnetos		Superseded by 96-12-07				UAE-2
81-18-04 R2	Oil pump		Superseded by 96-09-10				UAE-2
82-20-01	Bendix Magnetos		N/A- By Coupling Part#, Snap Ring Type Installed		X		UAE-2
82-27-03	Rajay Model 325E10 + 3AT6EE10J2 Turb charger - Turbine Housing Cracks - SL 27, Rev. A		N/A No Turbocharger Installed			X	UAE-2
87-10-06 R1	Rocker arm assemblies		N/A by Rockerarm Part #'s		X		UAE-2
89-04-02	Facet carburetors		N/A by Carburetor Model #		X		UAE-2
90-04-06 R1	Prop governor oil line		N/A Has Fixed Pitch Propeller		X		UAE-2
91-14-22	Crankshaft gear retaining bolt		Superseded by 2004-10-14				UAE-2
92-12-05	Piston pin failure		N/A by Piston Pin Part #		X		UAE-2
93-05-21	AC, Textron, Rajay fuel pumps		Superseded by 93-11-11				UAE-2
93-11-11	AC, Textron, Rajay fuel pumps		N/A no Fuel Pump Installed		X		UAE-2
93-18-03	Precision carburetors		Superseded by 98-01-06				UAE-2
93-19-04	Precision carburetors		N/A Carburetor Has Blue Epoxy Float		X		UAE-2
94-01-03 R2	Teledyne/Bendix magnetos		Found Embodied		X		UAE-2
94-06-09	Teledyne/Bendix magnetos		N/A by Capacitor Part #		X		UAE-2
94-14-13	Low octane detonation		Superseded by 95-26-02				UAE-2
95-26-02	Low octane detonation		N/A by Engine Overhaul Date		X		UAE-2
96-09-10	Oil pumps		Found Embodied		X		UAE-2
96-12-07	Prevent magneto failure and subsequent engine failure		Superseded by 2005-12-06				UAE-2
97-01-03	Piston pin		Superseded by 97-15-11				UAE-2
97-15-11	Piston pin		N/A by Piston Pin Part #		X		UAE-2
98-01-06	Precision carburetors		Found Embodied			X	UAE-2
98-02-08	Crankshaft corrosion		PID'ed			X	UAE-2
98-17-11	Crankshafts		N/A by Engine Overhaul Date		X		UAE-2
2004-10-14	CORRECTION - Prevent loosening or failure of the crankshaft gear retaining bolt		Embodied			X	UAE-2
2005-12-06	Prevent failure of the magneto impulse coupling assembly		Embodied L/H Magneto X-Values= 39 & 39 R/H Magneto N/A by Magneto Model #			X	UAE-2
2005-26-10	To prevent loss of engine power due to cracks in the cylinder assemblies		Superseded by 2006-12-07			X	UAE-2
2006-12-07	To prevent loss of engine power due to cracks in the cylinder assemblies		N/A by Cylinder Manufacturer			X	UAE-2
2007-04-19 R1	Cylinder separation		N/A by Cylinder Manufacturer				UAE-2
2008-19-05	To prevent loss of engine power due to cracks at the head-to-barrel interface in the cylinder assemblies and possible engine failure caused by separation of a cylinder head		Superseded by 2009-26-12			X	UAE-2
2009-26-12	To prevent loss of engine power due to cracks at the head-to-barrel interface and possible engine failure caused by separation of a cylinder head		N/A by Cylinder Manufacturer				UAE-2

Engine, Textron Lycoming, O-320 - 26 ADs.
Appliance, Carburetors, MA-4 - 7 ADs.
Appliance, Magnetos, TCM (formerly Bendix) S-20 Series - 8 ADs.

*=Recurring AD %=Superseded AD

- ✓ 65-02-02 - Drive adaptor gasket
- ✓ 69-10-07 - Cylinder baffle clamps
- ✓ 63-22-03 - Marvel Schebler carburetors
- ✓ 63-23-02 - Exhaust valve stem *
- ✓ 64-16-05 - Oil seal failure
- ✓ 66-20-04 - Oil filter adaptor gasket
- ✓ 69-24-03 - Marvel Schebler carburetors
- ✓ 72-06-05 R2 - Marvel Schebler carburetors
- ✓ 73-07-04 - Bendix Magnetos (Superseded by 94-01-03 R2) %
- ✓ 73-23-01 - Piston pin failures
- ✓ 74-26-09 - Bendix magnetos
- ✓ 75-08-09 - Oil pump
- ✓ 78-09-07 R3 - Bendix magnetos (Superseded by 96-12-07) % *
- ✓ 81-18-04 R2 - Oil pump (Superseded by 96-09-10) %
- ✓ 82-20-01 - Bendix magnetos
- ✓ 87-10-06 R1 - Rocker arm assemblies
- ✓ 89-04-02 - Facet carburetors
- ✓ 90-04-06 R1 - Prop governor oil line
- ✓ 91-14-22 - Crankshaft gear retaining bolt (Superseded by 2004-10-14) %
- ✓ 92-12-05 - Piston pin failure
- ✓ 93-05-21 - AC, Textron, Rajay fuel pumps (Superseded by 93-11-11) %
- ✓ 93-11-11 - AC, Textron, Rajay fuel pumps
- ✓ 93-18-03 - Precision carburetors (Superseded by 98-01-06) %
- ✓ 93-19-04 - Precision carburetors
- ✓ 94-01-03 R2 - Teledyne/Bendix magnetos
- ✓ 94-06-09 - Teledyne/Bendix magnetos
- ✓ 94-14-13 - Low octane detonation (Superseded by 95-26-02) %
- ✓ 95-26-02 - Low octane detonation
- ✓ 96-09-10 - Oil Pumps
- ✓ 96-12-07 - Prevent magneto failure and subsequent engine failure (Superseded by 2005-12-06) % *
- ✓ 97-01-03 - Piston pin (Superseded by 97-15-11) %
- ✓ 97-15-11 - Piston pin
- ✓ 98-01-06 - Precision carburetors *
- ✓ 98-02-08 - Crankshaft corrosion *
- ✓ 98-17-11 - Crankshafts
- ✓ 2004-10-14 - CORRECTION - Prevent loosening or failure of the crankshaft gear retaining bolt *
- ✓ 2005-12-06 - Prevent failure of the magneto impulse coupling assembly *
- ✓ 2005-26-10 - To prevent loss of engine power due to cracks in the cylinder assemblies (Superseded by 2006-12-07) %
- ✓ 2006-12-07 - To prevent loss of engine power due to cracks in the cylinder assemblies
- ✓ 2008-19-05 - To prevent loss of engine power due to cracks at the head-to-barrel interface in the cylinder assemblies and possible engine failure by separation of a cylinder head (Superseded by 2009-26-12) % *
- ✓ 2009-26-12 - To prevent loss of engine power due to cracks at the head-to-barrel interface and possible engine failure caused by a cylinder head *

All airworthiness directives applicable to:
Manufacturer: AVCO LYCOMING
Model: O-320

			AD Record List		Service Bulletin or Reference	Repeat Insp
REV	Model	Ctry.	AD Number	AD Subject		
O-	320	US	2009-26-12	Cracks at the head-to-barrel interface in the cylinder assemblies	REFER TO THIS DIRECTIVE	REFER T AD
O-	320	US	2008-19-05	SUPERSEDED BY FAA AWD 2009-26-12		REFER T AD
O-	320	US	2007-04-19R1	Cylinder separation	SUPERIOR AIR PARTS B06-01, REV. E	REFER T AD
O-	320	US	2006-12-07	To prevent loss of engine power due to cracks in the cylinder assemblies and possible engine failure caused by separation of a cylinder head.	05-08 REVISION 2	REFER T AD
O-	320	US	2004-10-14	CRANKSHAFT GEAR RETAINING BOLT	MSB 475C	REFER T AD
O-	320	US	75-08-09	OIL PUMP DRIVE SHAFT/IMPELLER	381B, 385C	NO
O-	320	US	95-26-02	TEXTRON LYCOMING ENGINES - SPECIFIC INSPECTIONS	(SB) NO. 398 AND (SI) NO. 1191	NO
O-	320	US	73-23-01R4	PISTON PINS	367F	REFER T AD
O-	320	US	96-09-10	ENGINE OIL PUMP IMPELLERS	454B, 455D, 524, 456	REFER T AD
O-	320	US	98-02-08	CRANKSHAFT INSPECTION	505B AND 530	REFER T AD
O-	320	US	98-01-06	MODEL MA-3,-3A,-3PA,-3SPA, MA4-SPA CARBURETORS INSP OF PRIMARY VENTURI -SB MSA-2R1	PRECISION AIRMOTIVE SB NO. MSA-2, REVISION 1	REFER T AD
O-	320	US	66-20-04	OIL FILTER ADAPTER GASKET	307	REFER T AD
O-	320	US	82-27-03	RAJAY MODEL 325E10 + 3AT6EE10J2 TURBOCHARGER - TURBINE HOUSING CRACKS- SL 27 REV A		REFER T AD

AIRCRAFT WEIGHT AND BALANCE REPORT

Aircraft over 3000 lbs. gross weight must be re-weighed every five years. All aircraft must be re-weighed immediately when alterations have resulted in an estimated 2% change in the empty weight either from a single change or an accumulation of changes.

1. AIRCRAFT IDENTIFICATION

Manufacturer: Piper Aircraft Corp.
Model: PA-22-150
Serial Number: 22-3388
Registration: C-FZNR
Name of Operator: Wallace Birch
Address: Box 13 Grp 341 RR 3
Winnipeg MB R3C 2E7

2. PERMISSABLE LIMITS: (from A/C Specs)

	Gross Weight	CofG Limits
Wheels:	2000 lbs	+17.5" to +23"
	1800 lbs	+12.0" to +23"
	1400 lbs	+ 9.5" to +23"
	or less	

3. **WEIGHING DATA:** Aircraft should be weighed with all required, optional or special equipment installed, full hydraulic and de-icing fluid and residual fuel and oil. If aircraft is not weighed empty, use space below to delete items installed but not included in empty weight, or to add items which are installed, but which should be included in the empty weight.

Oil included: 4.7L (5 USqts.) = 9.2 lb

Datum Location: Wing Leading Edge

Type of Scales used: Platform

Aircraft weighed on wheels with fairings

If aircraft is weighed on skis or floats, list the applicable installation drawings below:

Installation Drawings: _____

	Gross Wt	Tare Wt	Net Wt	Arm	Moment
Left Scale	400	0	400	31.5	12600.0
Right Scale	416	0	416	31.5	13104.0
Front Scale	348	0	348	-36.0	-12528.0

Empty Weight: 1164 lbs

Total Moment: 13176 in/lb

Empty weight center of gravity = $\frac{\text{Total Moment}}{\text{Empty Weight}} = 11.32 \text{ ins}$

Note: If center of gravity is outside of permissible limits, additional calculations should be included on a separate page to show that the center of gravity of the aircraft, when loaded in the most critical configuration can be brought within permissible limits. If fixed ballast is required to bring the center of gravity within limits this ballast should be included in the equipment list.

"I certify that this data has been prepared in accordance with the provisions of the Engineering and Inspection Manual and to the best of my knowledge represents the true empty weight and center of gravity of this aircraft."

Weighed at: CYAN

Date: JUNE 12-08

W. Birch PA36082
Signature and License Number of AME

Flight Load Factors
Max. Positive
Max. Negative
Airplane Loading

3.8
No Inverted Maneuvers Approved
Max. Wgt. (Take-Off and Landing)
2000 Pounds

C. G. Range

(Aft Wing Leading Edge)
{+17.5"} to {+23.0"} at 2000 lbs.
{+12.0"} to {+23.0"} at 1800 lbs.
{+9.5"} to {+23.0"} at 1400 lbs. or less

Maximum Baggage
Allowed

100 Pounds

Note: It is the responsibility of
the airplane owner and the
Pilot to insure that the air-
plane is properly loaded. (See
Weight and Balance.

Placards:

- (a) On the instrument panel in full
view of the Pilot:
 - (1) "Operate in Normal Category
in compliance with the Approved
Flight Manual. Acrobatics
(including spins) prohibited."
- (b) On the Baggage Compartment:
 - (1) "Maximum Baggage 100 Pounds."

Maneuvers

- (a) No acrobatic maneuvers approved
for Normal Category Operation.

Airspeed
Instrument
Markings
And Their
Significance

- (a) Radial Red line marks the never
exceed speed which is the maximum
safe airspeed 170 MPH. (148 Knots)
- (b) Yellow Arc on indicator denotes
range of speed in which operations
should be conducted with caution
and only in smooth air 135 to 170
MPH (117.0 to 148.0 Knots)
- (c) Green Arc denotes normal operating
speed range 53 to 135 MPH (39 to
117 Knots)
- (d) White Arc denotes normal operating
speed range with flaps extended
49 to 95 MPH (43 to 82 Knots)

JOE SKAVINSKI, DATE
WGM051818

OPERATION LIMITATION

Model PA-22-150, Serial No. 22-2624, 4 PCLM (Normal Category), Approved September 3, 1954.

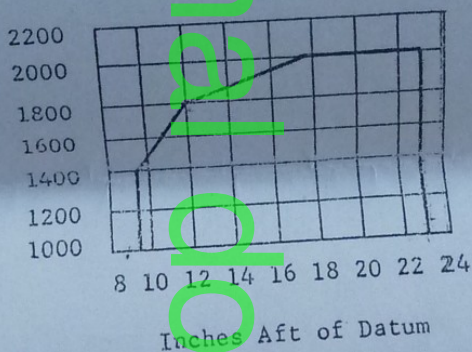
ENGINE
FUEL
ENGINE LIMITS
AIRSPEED LIMITS (CAS)

Lycoming O-320-A2B
80/87 Minimum grade aviation gasoline
For all operations, 2700 r.p.m. (150 hp.)
V_{ne} (never exceed) 170 m.p.h. (148 knots)
V_{no} (max. structural cruising) 135 m.p.h. (117 knots)
V_p (maneuvering) 112 m.p.h. (97 knots)
V_{fe} (flaps extended) 95 m.p.h. (82 knots)

C. G. range

Normal: (+17.5) to (+23.0) at 2000 lb.
(+12.0) to (+23.0) at 1800 lb.
(+ 9.5) to (+23.0) at 1400 lb. or less

GROSS WEIGHT
(lb.)



EMPTY WEIGHT
MAXIMUM WEIGHT
NO. SEATS
MAXIMUM BAGGAGE
FUEL CAPACITY
OIL CAPACITY
CONTROL SURFACE MOVEMENTS

Normal: 1136 lb.
Normal: 2000 lb.
4 (2 at +21, and 2 at +49).
100 lb. (+67)
36 gal. (2 Wing tanks at +24) (216 lb)
2 gal. (-29) (15 lb)
Stabilizer 1° Up 6 1/2° Down
Elevator 24° Up 12° Down
Aileron 15° Up 15° Down
Rudder 16° Right 16° Left
Flap 40° Down

JOE SKAVINSKI, DATE
WGM051818

C-FZNR

Equipment List

June 12 / 08

C-FZNR		Equipment List			June 12 / 08			REVISED	REVISED	REVISED	
DATE	ITEM	WEIGHT	ARM	MOMENT	EMPTY WT	MOMENT	C of G	INITIALS			
	Propeller, Sensenich M74DM	30	-50	-1500							
	Spinner	2	-52	-104							
	Oil Cooler	6	-46	-104							
	Vacuum Pump	4	-25	-100							
	Starter (Lamar)		-40								
	Two Main Wheel Brake Assemblies 6.00 - 6	14	31.5	441							
	Two Main 4-Ply Rating Tires, 6.00 - 6 with Regular Tubes	17	31.5	535.5							
	One Nose Wheel 6.00 - 6 Cleveland Wheel Assembly	5	-36	-180							
	One Nose Wheel 4-Ply Tire 6.00 - 6 with Regular Tube	9	-36	-324							
	Two Landing Lights	4	5	20							
	Battery	12	21	252							
	Generator (Alternater)		-40								
	Navigation Lights	3	65	195							
	Fire Extinguisher	2	54	108							
May 15 / 03	ELT AK-450	3.1	60	186							
	Radio Comm 120	2.5	0	0							
	Radio Comm 11A	2.5	0	0							
	2 Whip Antennas	0.4	15	6							
Sept 01 / 03	Transponder, ARC RT-359A	3.5	0	0							
May 14 / 01	Alt Encoder, Terra 3000	0.4	-20	-8							
	Intercom, Hypervox HV 1	0.5	0	0							
Dec 21 / 07	Two Main Wheel Fairings	14	32	448							
Dec 21 / 07	One Nose Wheel Fairing	6.1	-32	-195.2							

THIS DOCUMENT MUST BE KEPT IN THE AIRPLANE AT ALL TIMES

CAA APPROVED
APPROVAL BASIS CAR 3 AND 410
AUGUST 27, 1955
NORMAL CATEGORY
2000 POUNDS GROSS WEIGHT

A.A. Identification No. C-FZNR

JOE SKAVINSKI, DATE

AIRPLANE FLIGHT MANUAL

WGM051818

Limitations

The following limitations must be observed in the operation of this airplane:

Engine	Lycoming O-320-A2B 150 HP
Engine Limits	For all operations 2700 RPM
Fuel	80/87 Octane Minimum Aviation Gasoline
Propellers	(a) Sensenich M74DM, Fixed Pitch Metal
	74.0" Maximum Diameter
	72.0" Minimum Diameter
	Static Limits: Maximum 2450 RPM
	Minimum 2250 RPM
Power Instruments	Oil Temperature-Unsafe if indicator exceeds
	Red line (245 degrees F.)
	Yellow Arc: Caution (40 degrees F. to
	120 degrees F.)
	Green Arc: Normal Operating Range
	(120 degrees F. to 245 degrees F.)
	Oil Pressure-Unsafe if indicator exceeds
	Red line (100 lbs.) or is below
	the Red line (25 lbs. minimum)
	Yellow Arc: Caution (85 lbs. to 100 lbs.)
	and (25 lbs. to 65 lbs.)
	Green Arc: Normal Operating Range
	(65 lbs. to 85 lbs.)
	Tachometer-Red line: Rated Engine Speed
	Green Arc: 500 RPM to 2700 RPM
	Normal Operating Range
Flap Position	Take-Off 0 Degrees Landing 40 Degrees
Airspeed Limits	
(True Ind. Airspeed)	Normal Category
Maneuvering	112 MPH 97.0 Knots
Max. Cruising Speed	135 MPH 117.0 Knots
Never Exceed	170 MPH 148.0 Knots
Flaps Extended	95 MPH 82.0 Knots

JOE SKAVINSKI, DATE
WGM051818

1-800-463-6377
NAV CANADA

CANADIAN FLIGHT PLAN AND FLIGHT ITINERARY
PLAN DE VOL ET ITINÉRAIRE DE VOL CANADIEN

ICAO FLIGHT PLAN
PLAN DE VOL OACI

PRIORITY / PRIORITÉ

<< = FF →

ADDRESSEE(S) / DESTINATAIRE(S)

FILING TIME / HEURE DE DÉPÔT

ORIGINATOR / EXPÉDITEUR

SPECIFIC IDENTIFICATION OF ADDRESSEE(S) AND/OR ORIGINATOR / IDENTIFICATION PRÉCISE DU(DES) DESTINATAIRE(S) ET/OU DE L'EXPÉDITEUR

3 MESSAGE TYPE
TYPE DE MESSAGE

<< = (FPL

7 AIRCRAFT IDENTIFICATION /
IDENTIFICATION DE L'AÉRONEF

= FZNR

8 FLIGHT RULES /
RÈGLES DE VOL

= V YOE

TYPE OF FLIGHT /
TYPE DE VOL

= G

9 NUMBER / NOMBRE

=

TYPE OF AIRCRAFT / TYPE D'AÉRONEF

= PA.2.2

WAKE TURBULENCE CAT. /
CAT. DE TURBULENCE DE SILLAGE

=

10 EQUIPMENT / ÉQUIPEMENT

= VG/C

13 DEPARTURE AERODROME / AÉRODROME DE DÉPART

= CYAV

TIME / HEURE

= 1200

15 CRUISING SPEED /
VITESSE DE CROISIÈRE

= N0090

ALTITUDE / LEVEL / NIVEAU

= A025

ROUTE / ROUTE

= YQV (0045) YQV YXE YXE

18 DESTINATION AERODROME /
AÉRODROME DE DESTINATION

= CYXE

TOTAL EET / DURÉE TOTALE ESTIMÉE

= 0545

SAR

=

ALTN AERODROME /
AÉRODROME DE DÉGAGEMENT

=

2ND ALTN AERODROME /
2e AÉRODROME DE DÉGAGEMENT

=

19 OTHER INFORMATION / RENSEIGNEMENTS DIVERS

=

19 ENDURANCE / AUTONOMIE

= E / 0345

PERSONS ON BOARD / PERSONNES À BORD

→ P / 1

EMERGENCY RADIO / RADIO DE SECOURS

→ R / V

VHF

ELT

ELT TYPE / TYPE D'ELT

= A, F

SURVIVAL EQUIPMENT / ÉQUIPEMENT DE SURVIE

→ S / P

JACKETS / GILETS DE SAUVETAGE

→ J / L

UHF

VHF

DINGHIES / CANOTS

→ D /

COLOUR / COULEUR

=

NUMBER / NOMBRE

→ D /

CAPACITY / CAPACITÉ

=

AIRCRAFT COLOUR AND MARKINGS / COULEUR ET MARQUES DE L'AÉRONEF

= BLUE ON WHITE

REMARKS / REMARQUES

=

AN ARRIVAL REPORT WILL BE FILED WITH / UN COMPTE RENDU D'ARRIVÉE SERA NOTIFIÉ À

= YXE

NAME AND PHONE NUMBER OR ADDRESS OF PERSON(S) OR COMPANY TO BE NOTIFIED IF SEARCH AND RESCUE ACTION INITIATED /
NOM ET NUMÉRO DE TÉLÉPHONE OU ADRESSE DE LA (DES) PERSONNE(S) OU COMPAGNIE À AVISER SI DES RECHERCHES SONT ENTREPRISES

= DORDEN BIRCH 204-661-1821

PILOT-IN-COMMAND / PILOTE COMMANDANT DE BORD

= W. BIRCH

PILOT'S LICENCE NO. / N° DE LICENCE DU PILOTE

= PA36082

FILED BY / DÉPOSÉ PAR

SPACE RESERVED FOR ADDITIONAL REQUIREMENTS / ESPACE RÉSERVÉ À DES FINS SUPPLÉMENTAIRES

UNIVERSAL AERO ENGINES LTD.

421 AIRPORT DRIVE
WPG./ST. ANDREWS AIRPORT
ST. ANDREWS, MB
R1A 3R2

Phone: 339-4078

FAX: 338-2230

NEW PARTS INSTALLED

MAKE: Lycoming

MODEL: O-320

SERIAL #: L-5765-27

DATE: MAY 3 1999

WORK ORDER: 871

QTY	PART #	DESCRIPTION	QTY	PART #	DESCRIPTION
4	71577	TUBE	1	61155	GEAR (SERV)
8	72877	LINKER BODY	1	78531	OIL PUMP BODY
4	67447 m10	BEARING			MISC HARDWARE
2	13884 m10	BEARING			BOLTS, WASHER BUSHINGS
8	61662 m10	ROD BEARING			SEALS
8	12186	ROD NUT	1	46-110	VENTURI
1	69371-1	GASKET SET	1	30-766	FLOAT
1	18109 AS	OIL PUMP KIT	1	71167R	CAMSHAFT
4	75413 P10	PISTONS			
4	36015C P10	RING SET			
4	74230	GUIDE			
4	13890	ROCKERSHAFT			
8	60828	PISTON PIN PLUG			
8	12872	TEFLON BUTION			
1	16B-42	CARB KIT			
4	57B 1821	HOSE			
4	69603	HOSE			
1	10-357174	POINT			
1	10-51676	CAPACITOR			
1	10-160844	CARBON BRUSH			
1	10-357592	OIL SEAL			
1	57A-2213	BOLT			

I certify I have inspected this aircraft in compliance with the Condition and Conformity Inspection procedure prescribed in the Engineering and Inspection Manual and it is airworthy.

E. H. [Signature] 20m 684

I hereby certify that I have completed the nearest equivalent to a 100-hour periodic inspection of aircraft described herein and certify the aircraft is airworthy.

JOE SKAVINSKI, DATE *Aug 10/90*

Sig. AMF *[Signature]*

Date *WGM051818*

AR Blain MO 67591

AR Blain MO 67591

W. Scully PA 36082

N. Scully

inspection carried out certified airworthy

inspection carried out certified airworthy

INTERVANCE CATEGORY

MAXIMUM HOURS BETWEEN OVERHAULS _____ HOURS.

OR LIFE

Aug 10/90	88	502.00		
Aug 10/90		549.0	100 hr annual inspection	carried out
July 1999		573.2	annual inspection	carried out
JULY 21-00		631.5	CHANGED TO OWNER MAINTENANCE	CATEGORY
AUG. 4-00		631.5	ANNUAL INSPECTION	
CARRIED FORWARD				
TOTAL THIS PAGE				
TOTAL FROM PREVIOUS SUMMARY				
TOTAL SINCE MFG.				

MAXIMUM HOURS

July 18/90	3	2766 D	
July 30/90	5	2766 5	
Aug 10/91			
Aug 10/91			100 hr annual inspect
17-91			new seal installed in new lower heat hoses all AD checked
July 10/99	2766.5	aircraft re assembled installed modified wing struts per STC SA4635 NM	after repairs - fuselage # compass checked I see work certified aircraft successful
CARRIED FORWARD			

Replane Bm
 Ben filter
 Ben Craft 1000
 scale on aug 2

Carried out certifica
 subject to
 nose wheel steering
 installed

45	2395.30		
00	2396.30		
30	2398.00		
30	2399.30		
00	2403.30		
30	2408.00		
30	2409.30		
30	2412.00		
00	2413.00		
2457	5	Wings and tail section removed. The wings & tail checked for addition, checked and ground after Aircraft stripped of old fabric. Tail section recovered with Dacron fabric - style D 101A, 3-7 oz per sq. ft. Finished with the old fabric aircraft covering process, Poly first and Poly spray. Painted with Dupont base: wings and fuselage - DuPont white "10. DuPont: DuPont Red #100 and "10. Aircraft assembled and type inspected. Registration changed from CF-ZNR to C-FZNR. All controls checked for correct routing, R & P control LOCH SHARK 7K FLT installed. movement, and security. Date SIGNATURES: E. Alfpo 100 HR insp carried out for C of A renewal. Kit changed - shell 65 W. Kit and fuel system checked & changed. 2 1/2 liter Wheel turning fluid. Battery serviced. New 100.26 tube installed in nose wheel	4/11/13 4/11/13 4/11/13 4/11/13 4/11/13 4/11/13 4/11/13 4/11/13 4/11/13 4/11/13
2458		TEST FLIGHT	
CARRIED FORWARD			

I hereby certify I have completed the nearest equivalent
 to a ... Hr. periodic inspection of the aircraft de-
 scribed herein and it is airworthy.
 I hereby certify I have flown this aircraft and its perform-
 ance, flying qualities, functions of controls, power plant
 and landing gear, etc. were equivalent to the standard of
 the type.

E. Alfpo
 4/11/13

[illegible][illegible]

1976				2488	
Jan.	3	3	30	2489	
Feb	8	1	00	2489.5	
April	6		30	2490	
June	4		30	2493	
	8	3	00	2493.5	
	14		30	2496.5	
	15	3	00	2500.1	
July	16	3	30	2500.5	
	17		30	2501	
	18		30	2505.5	
Sept	6	4	30	2535	
Sept. 4/76	-	-	-	2535.5	100 HA base for CC I
April 27/77				2530 0	A.D. Ammendment 39-2833
	28	1		2531	
MAY 1	3	30		2539.5	
June 27			30	2535	

CARRIED FORWARD

MAKE AND MODEL 120	SERIAL NOS. 12-5765-272	TIME SINCE OVERHAUL 1 229.0	2
MAKE AND MODEL 174 D158	SERIAL NOS. 1 19098	TIME SINCE OVERHAUL 1 268.0	2
LAST C.C.I. IT HAS BEEN ED ON:	<input type="checkbox"/> SKIS <input type="checkbox"/> FLOATS	MANUFACTURER	
REGISTERED OWNER IS PRIVATE OPERATOR AND IT IS OR HAS BEEN LEASED TO COMMERCIAL OPERATOR(S) ENTER HOURS FLOWN BY COMMERCIAL OPERATOR(S)			
I CERTIFY I HAVE INSPECTED THIS AIRCRAFT IN COMPLIANCE WITH THE CONDITION AND CONFORMITY INSPECTION PROCEDURE PRESCRIBED IN THE ENGINEERING AND INSPECTION MANUAL AND THE AIRCRAFT IS AIRWORTHY OR SERVICEABLE*, WHICHEVER IS APPLICABLE. NOTE: FOR ULTRA-LIGHT AIRCRAFT THIS CERTIFICATION MAY BE SIGNED BY THE OWNER IN LIEU OF AN AME OR APPROVED INSPECTOR APPLIES TO AIRCRAFT ISSUED WITH A FLIGHT PERMIT.			
SIGNATURE OF AME OR APPROVED CO. INSPECTOR Elgin Help		AME/CO. NO. DATE x DP12693 8 SEPT 76	

Service Bull. #528.

Transport Canada	Transport Canada	AIRCRAFT CONDITION AND CONFORMITY INSPECTION (C.C.I.) REPORT				DATE OF ISSUE OF COF/FLIGHT PERMIT DAY 8 MO. SEPT. YR.
UNLAWFULITY REGISTRATION MARKS 3-FZNR	AIRCRAFT NORMALLY BASED AT THE PAS, MAN	REGION CENTRAL				
AIRCRAFT MAKE, MODEL AND TYPE APPROVAL/FAA CERT. NO. 120	SERIAL NO. 12-5765-272	SERIAL NO. 22-3399	TIME SINCE NEW 2505.5			AIRCRAFT ISSUED WITH: <input checked="" type="checkbox"/> CERTIFICATE OF AIRWORTHINESS <input type="checkbox"/> FLIGHT PERMIT
MAKE AND MODEL 120	SERIAL NOS. 1 19098	2	3	4		AIRCRAFT PURPOSE: <input checked="" type="checkbox"/> PRIVATE <input type="checkbox"/> COMMERCIAL
PELLER MAKE AND MODEL 174 D158	SERIAL NOS. 1 268.0	2	3	4		LIST AIRWORTHINESS DIRECTIVES AIRCRAFT SINCE LAST C.C.I. PROC CF 68-4
LAST C.C.I. AIRCRAFT HAS BEEN OPERATED ON:	<input type="checkbox"/> SKIS <input type="checkbox"/> FLOATS	MANUFACTURER	MODEL	MODEL		60-1-7 60-10-8
REGISTERED OWNER IS PRIVATE OPERATOR AND IT IS OR HAS BEEN LEASED TO COMMERCIAL OPERATOR(S) ENTER HOURS FLOWN BY COMMERCIAL OPERATOR(S)			CURRENT CALENDAR YEAR	LAST CALENDAR YEAR		AIRCRAFT TIME SINCE NEW 1ST JAN. THIS YEAR 2 1ST JAN. LAST YEAR 2
(1) I CERTIFY I HAVE INSPECTED THIS AIRCRAFT IN COMPLIANCE WITH THE CONDITION AND CONFORMITY INSPECTION PROCEDURE PRESCRIBED IN THE ENGINEERING AND INSPECTION MANUAL AND THE AIRCRAFT IS AIRWORTHY OR SERVICEABLE*, WHICHEVER IS APPLICABLE. NOTE: FOR ULTRA-LIGHT AIRCRAFT THIS CERTIFICATION MAY BE SIGNED BY THE OWNER IN LIEU OF AN AME OR APPROVED INSPECTOR. * APPLIES TO AIRCRAFT ISSUED WITH A FLIGHT PERMIT.			(2) I CERTIFY THAT I HAVE FLOWN THIS AIRCRAFT AND ITS PERFORMANCE QUALITIES, FUNCTIONS OF CONTROLS, POWERPLANT, AND LANDINGS WERE EQUIVALENT TO THE STANDARD OF THE TYPE. NOTE: THE ELAPSED TIME BETWEEN THE DATE OF FLIGHT CERTIFICATION (1) SHALL NOT EXCEED 7 CONSECUTIVE CALENDAR			
SIGNATURE OF AME OR APPROVED CO. INSPECTOR Elgin Help		AME/CO. NO. DATE x DP12693 8 SEPT. 76	DATE OF FLIGHT 8 SEPT. 76	LICENCE NO. x DP12693	SIGNATURE OF PILOT Elgin Help	

Service Bull. #528.

CC I
amendment 39-2833

Copy of original

ENGINE S
INSTALL
BOTH PAGES AND
DIFF. PRE
LEAD RISE
D. REGULA
D. PROP

July 10/99 639.9

UNIVERSAL AERO ENGINES LTD.
421 AIRPORT DRIVE, WINNIPEG, ST. ANDREWS, MANITOBA R1A 3R2

D.O.T. APPROVAL NO. **81-91**

Nomenclature: ALRO ENGINE

Manufacturer: LYCOMING Type/Model: O-320

Part No: _____ Serial No: 4-5765-27

Total Time: _____ Work Order: UAF 871

New ☐ Overhaul ☐ Repair ☒ Modification ☐

Previous certification: _____

The maintenance described has been performed in accordance with the applicable standards of airworthiness.

DATE MAY 3/99 NAME OR AUTHORIZED INSPECTOR [Signature] LIC. No./STAMP VI376750

MAINTENANCE RELEASE

ENGINE ACCORDANCE
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engine
new
certified
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CAT 660

UNIVERSAL AERO ENGINES LTD.
St. Andrews Airport Box 27, Group 6A, R.R.1 Winnipeg, Manitoba R3C 2E4 (204) 339-4078

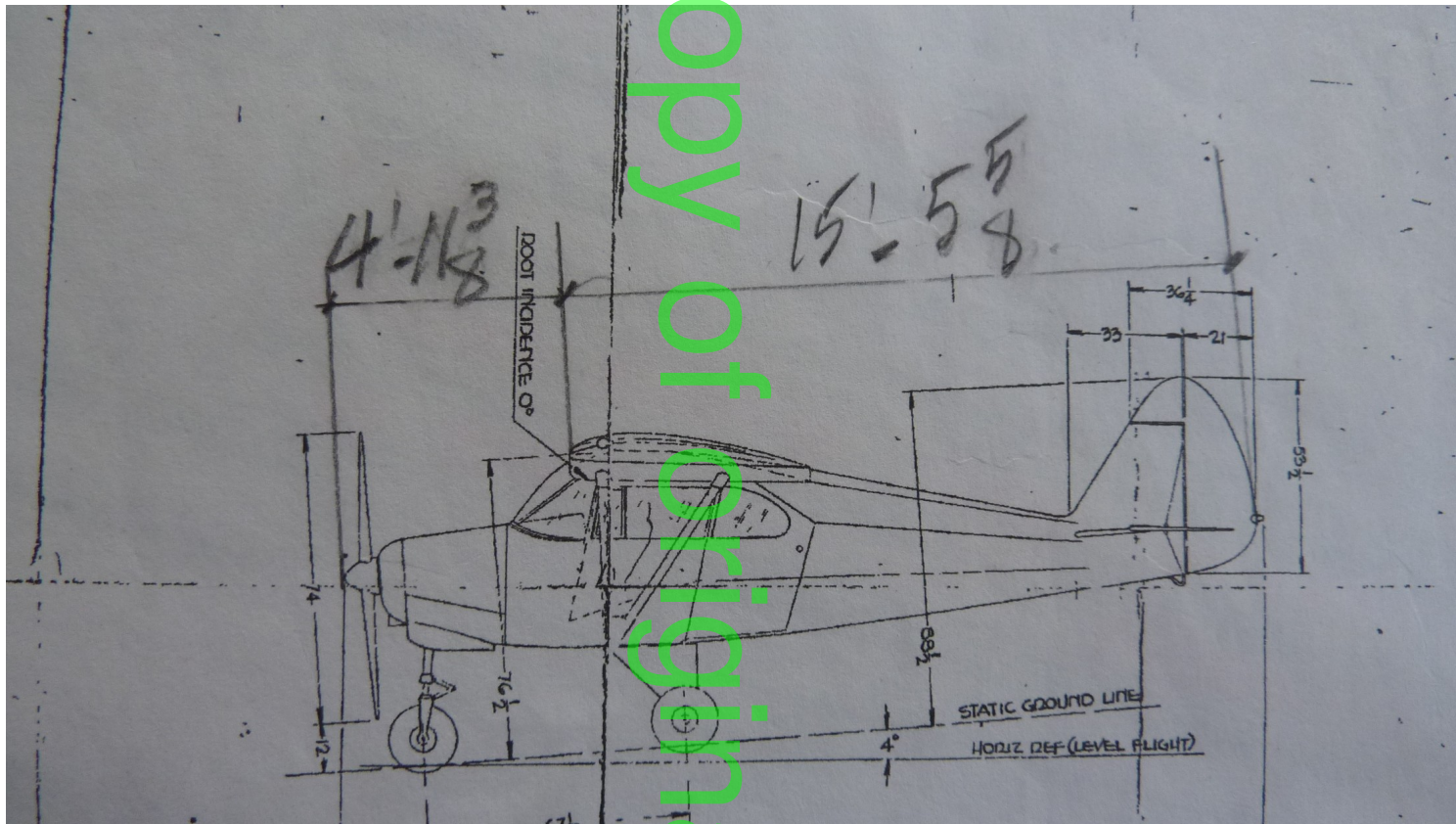
ENGINE TEST LOG

DATE MAY 3 1999 CUSTOMER WALLY BIRCH W.O. NO. 871 ENG. MODEL O-320 D.O. NO. 4-5765-27

AMB. AIR TEMP. 15°C BAROMETER _____

TIME START	TIME STOP	TOTAL TIME	RPM	MP	OIL PRESS	OIL TEMP	CYL TEMP	FUEL PRESS	HOT FUEL PRESS	FF	H.P.
10:00	10:15	15	1000								
10:15	10:20	5	1200								
10:20	10:25	5	1400								
10:25	10:30	5	1600								
10:30	10:35	5	1800								
ENGINE STOP CHECK OIL SCREEN & CHECK FOR LEAKS (FOUND OK)											
11:00	11:05	5	1000								
11:05	11:10	5	1400								
11:10	11:15	5	1800								
11:15	11:20	5	2000								
11:20	11:25	5	2400								
11:25	11:30	5	2600								
ENGINE STOP CHECK OIL SCREEN FOUND OK											
COMPRESSION TEST											
EGT											
1	2	3	4	5	6	7	8	79	78	77	76
								80	80	80	80
T/C# 5											
MENT 5											
DATE											
1400 LEFT 50 RIGHT 75											
REMARKS											

Form E Page 25

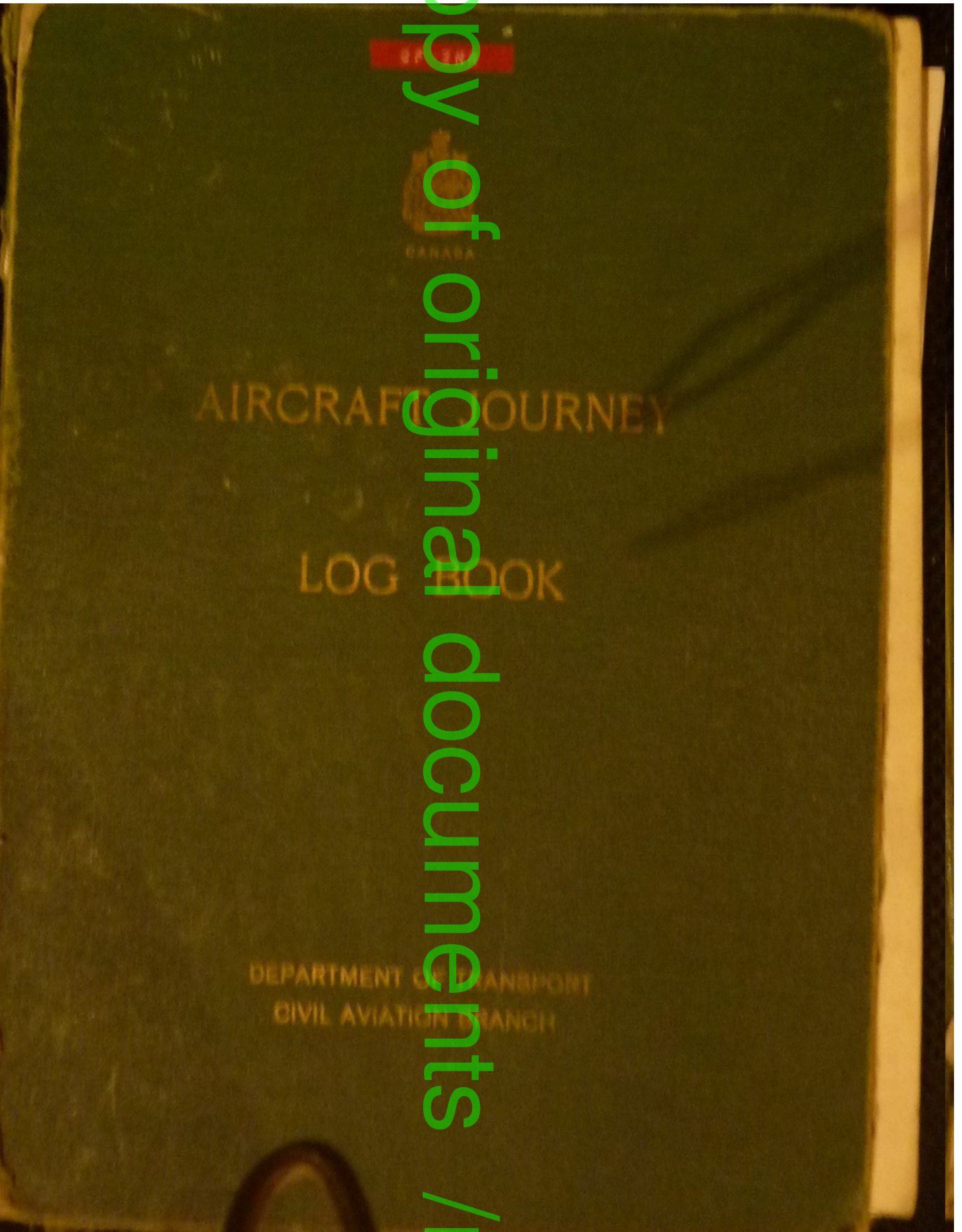


C-FZNR WEIGHT & BALANCE

	Weight (lb.)	Arm (in.)	Moment
Basic Empty Weight	1164	10.3	11989
Usable Fuel (136 L x 1.6#)	218	24	5232
Pilot & Front Passenger <i>395</i>	<i>330</i>	21	<i>8295</i> <i>6930</i>
Rear Passengers	<i>278</i>	49	<i>13622</i>
Baggage (100 lb max)	10	67	<i>670</i>
Total weight and moment	<i>2000</i>	<i>19.2</i>	<i>38443</i>

C. G. Range: 17.5 to 23 @ 2000 lb
 12 to 23 @ 1800 lb
 9.5 to 23 @ 1400 lb

copy of original documents /nf



AIRCRAFT

1. Nationality and Registration C-FZNR
~~CF-ZNR~~
2. Manufacturer's Designation PA-22 TRI-PACER
3. Type Approval or Specification Number 1A6
4. Manufacturer's Serial Number 22-3388 (1956)
5. Fuel - Capacity 30 US 30 GALS 1 MP. 136 L.
 - Grade 80/87 or HIGHER
6. Oil - Capacity 8 US 67 QTS. 7.6 L.
 - Grade - Summer E 100 Shell W 100
 - Winter E 80 W 80 } DETERGENT
7. Hydraulic Fluid Specification UNIVIS J43
8. Maximum Gross Weight Authorized NORMAN-2000th UTILITY 1680

EMPTY WT: 1142 lbs.

PROPELLER - SUNENICH M74DM-5860"

JOURNEY			CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS		SIGNATURES	
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE MFG.	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. SAGE CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER			
BROUGHT FORWARD								NO.	LBS.								
Aug 19	Requena	J. Tallafra	18:30	17:00	1:30	1:30	2252:35	2	360	20	645	200		1972			
Aug 20	Requena	J. Tallafra	15:00	15:35	1:35	1:35	2253:10	2	300	20	645	200		1972			
"	Requena	J. Tallafra	15:00	15:45	1:45	1:45	2254:58	2	300	20	645	200		1972			
Aug 21	Requena	J. Tallafra	15:55	16:05	1:10	1:10	2257:08	2	340	20	645	200		1972			
Aug 24	Requena	J. Tallafra	16:45	17:55	1:10	1:10	2258:18	2	340	22	645	200		1972			
Aug 24	Requena	J. Tallafra	20:00	20:22	1:22	1:22	2258:40	2	290	14	645	200		1972			
Aug 27	Requena	J. Tallafra	20:45	21:15	1:30	1:30	2259:10	3	440	30	645	200		1972			
Sept 3	Requena	J. Tallafra	17:30	18:30	1:00	1:00	2260:10	3	500	28	645	200		1972			
Sept 9	Requena	J. Tallafra	10:30	12:00	1:30	1:30	2261:30	2	300	30	645	200		1972			
Sept 10	Requena	J. Tallafra	10:30	12:00	1:30	1:30	2262:00	2	300	30	645	200		1972			
Sept 15	Requena	J. Tallafra	10:30	12:00	1:30	1:30	2262:30	2	300	30	645	200		1972			
Sept 17	Requena	J. Tallafra	17:00	17:45	1:45	1:45	2263:15	2	300	30	645	200		1972			
Sept 20	Requena	J. Tallafra	10:00	10:30	1:30	1:30	2263:45	2	300	30	645	200		1972			
Sept 20	Requena	J. Tallafra	11:00	11:45	1:45	1:45	2264:30	2	320	20	645	200		1972			
Sept 30	Requena	J. Tallafra	16:00	16:30	1:30	1:30	2265:00	2	300	30	645	200		1972			
Oct 1	Requena	J. Tallafra	11:30	12:10	1:40	1:40	2265:40	3	370	20	645	200		1972			
Oct 1	Requena	J. Tallafra	15:30	16:00	1:30	1:30	2267:10	2	300	30	645	200		1972			
Oct 6	QR-QR	J. Tallafra	16:30	17:30	1:00	1:00	2267:45	2	300	30	645	200		1972			
Oct 7	QR-Local	J. Tallafra	11:00	12:15	1:15	1:15	2268:00	2	300	30	645	200		1972			
Oct 7	QR-Local	J. Tallafra	16:00	17:00	1:00	1:00	2269:00	2	300	30	645	200		1972			
Oct 7	QR-Local	J. Tallafra	18:45	19:45	1:00	1:00	2270:00	2	300	30	645	200		1972			
Oct 8	QR-Local	J. Tallafra	9:45	11:00	1:15	1:15	2271:15	2	330	30	645	200		1972			
Oct 8	QR-Local	J. Tallafra	19:20	20:20	1:00	1:00	2272:15	2	330	30	645	200		1972			
Oct 9	QR-Local	J. Tallafra	11:00	12:00	1:00	1:00	2273:15	2	300	30	645	200		1972			
Oct 14	QR-Local	J. Tallafra	10:15	11:00	1:45	1:45	2274:00	3	460	20	645	200		1972			
Oct 21	QR-Local	J. Tallafra	11:00	11:45	1:45	1:45	2275:35	4	700	10	645	200		1972			
Oct 21	QR-Local	J. Tallafra	15:40	16:25	1:45	1:45	2276:20	2	380	30	645	200		1972			
Oct 27	QR-QR	J. Tallafra	19:35	20:40	1:05	1:05	2277:25	1	190	30	645	200		1972			
Oct 27	QR-Local	J. Tallafra	18:45	19:35	1:50	1:50	2278:15	1	190	30	645	200		1972			
TOTAL THIS PAGE																	

3

JOURNEY		CREW		RECORD OF TIME			
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE MFG.
BROUGHT FORWARD →							
Dec 27/72	Requena	J. Tallafra	14:30	16:00	1:30	1:30	2279:45
Jan 13	Requena	J. Tallafra	13:30	14:45	1:15	1:15	2280:00
Jan 22	QR-Local	J. Tallafra	16:30	17:30	1:00	1:00	2281:00
Jan 27	QR-Local	J. Tallafra	10:45	11:35	1:50	1:50	2282:50
Feb 27	QR-Local	J. Tallafra	17:45	18:30	1:45	1:45	2284:35
Mar 1	QR-Local	J. Tallafra	9:00	10:15	1:15	1:15	2285:50
Mar 1	QR-Local	J. Tallafra	14:30	15:10	1:40	1:40	2287:30
Mar 1	QR-Local	J. Tallafra	16:30	17:00	1:30	1:30	2289:00
Mar 2	QR-Local	J. Tallafra	9:00	9:45	1:15	1:15	2290:15
Mar 2	QR-Local	J. Tallafra	21:00	21:25	1:25	1:25	2291:40
Mar 3	QR-Local	J. Tallafra	13:45	14:15	1:00	1:00	2292:40
Sept 11	QR-Local	J. Tallafra	11:40	12:30	1:50	1:50	2294:30
Sept 11/73	QR-Local	J. Tallafra	18:45	19:45	1:00	1:00	2295:30
Sept 11/73	QR-Local	J. Tallafra	19:00	19:40	1:20	1:20	2296:50
Sept 11/73	QR-Local	J. Tallafra	19:45	20:15	1:15	1:15	2298:05
Sept 11/73	QR-Local	J. Tallafra	20:45	21:45	1:00	1:00	2299:05
Sept 11/73	QR-Local	J. Tallafra	13:20	14:20	1:00	1:00	2300:05
Sept 11/73	QR-Local	J. Tallafra	10:30	11:15	1:45	1:45	2301:50
Sept 11/73	QR-Local	J. Tallafra	10:20	11:35	1:35	1:35	2303:25
Sept 11/73	QR-Local	J. Tallafra	10:35	11:25	1:00	1:00	2304:25
Sept 11/73	QR-Local	J. Tallafra	10:50	11:40	1:00	1:00	2305:25
Sept 30/73	QR-Local	J. Tallafra	17:05	17:55	1:45	1:45	2307:15
Sept 30/73	QR-Local	J. Tallafra	10:45	11:30	1:15	1:15	2308:30

RECORD OF WEIGHTS					REMARKS		SIGNATURES	
9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. SAGE CARGO	13. TOTAL WEIGHT AT T/O	DEFECTS & AIRWORTHINESS CERTIFICATION	PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER		
NO.	LBS.							
2	360	30	645	200				
1	180	30	645	200				
2	350	30	645	200				
1	200	30	645	200				
2	320	30	645	200				
2	350	30	645	250				
2	350	30	645	250				
2	350	25	645	250				
2	350	30	645	250				
1	190	30	645	400				
2	350	25	645	400				
2	350	20	645	400				
1	175	30	645	200				
2	240	15	645	1500				
1	175	30	645	1500				
2	225	10	645	1500				
4	700	15	645	1500				
1	175	30	645	1000				
3	500	20	645	1000				
1	175	30	645	1000				
4	425	20	645	1000				
1	175	15	645	1000				
4	500	30	645	1000				

I hereby certify that I have this day inspected this aircraft in compliance with the Condition and Conformity Inspection procedure prescribed in the Engineering and Inspection Manual, and the aircraft is airworthy.

Signed: *[Signature]* Lic. No. *[Number]*

RECORD OF WEIGHTS						REMARKS	SIGNATURES				
8.	TOTAL PERSONS ON BOARD	9.	FUEL, LBS. GAL.	11.	WEIGHT EQUIP. BAG, CASE CARGO	12.	TOTAL WEIGHT AT T/O	13.	DEFECTS & AIRWORTHINESS CERTIFICATION	14.	PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
	NO.	LBS.									
	1	135	30	550	—	1550					W. J. [Signature]
	4	670	25	500	—	1350					W. J. [Signature]
	1	125	15	500		1450					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	3	400	25	500	10	1660					W. J. [Signature]
	1	125	20	500	10	1500					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	1	125	20	500	10	1500					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	1	125	25	500	10	1525					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	1	125	25	500	10	1525					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	1	125	20	500	10	1500					W. J. [Signature]
	1	125	20	500	10	1500					W. J. [Signature]
	1	125	10	500	10	1415					W. J. [Signature]
	3	400	30	500	10	1620					W. J. [Signature]
	3	400	35	500	10	1660					W. J. [Signature]
	1	180	30	500	10	1560					W. J. [Signature]
	3	350	25	500	10	1610					W. J. [Signature]
	4	200	30	500	10	1550					W. J. [Signature]
	100	1000	500	500	CR	2000					W. J. [Signature]
	REMARKS										W. J. [Signature]
	3	360	30	500	10	1610					W. J. [Signature]
	1	125	15	500	10	1530					W. J. [Signature]
	1	125	10	500	10	1500					W. J. [Signature]
	1	125	30	500	10	1560					W. J. [Signature]
	3	360	25	500	10	1630					W. J. [Signature]
	1	120	30	500	25	1580					W. J. [Signature]
	3	360	30	500	25	1650					W. J. [Signature]

RECORD OF WEIGHTS						REMARKS	SIGNATURES
B.	TOTAL PERSONS ON BOARD	9.	10.	11.	12.	13.	14.
NO.	LBS.	FUEL IMP. GAL.	OIL IMP. GAL.	WEIGHT EQUIP. BAG-GAGE CARGO	TOTAL WEIGHT AT T/O	DEFECTS & AIRWORTHINESS CERTIFICATION	PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
1	180	20	60	25	1580		10/13 WEF 1018
1	180	20	70	20	1600		10/13 "
4	400	20	60	25	1800		10/13 "
2	330	10	60	25	1160		10/13 "
1	180	20	60	35	1450		10/13 "
2	355	25	50	25	1680		10/13 "
1	180	20	60	35	1560		10/13 "
1	180	20	60	25	1560		10/13 "
1	180	15	60	25	1450		10/13 "
4	430	20	60	25	1780		10/13 "
1	180	25	60	25	1585		10/13 "
1	180	22	60	35	1500		10/13 "
1	180	20	60	25	1500		10/13 "
4	380	15	50	25	1665		10/13 "
1	180	20	60	25	1560		10/13 "
2	320	30	60	80	1760		10/13 "
4	320	20	60	100	1800		10/13 "
1	180	20	60	10	1500		10/13 "
2	360	25	60	10	1450		10/13 "
3	420	20	60	10	1250		10/13 "
1	180	25	60	10	1580		10/13 "
1	180	20	60	10	1520		10/13 "
1	180	25	60	10	1800		10/13 "
1	180	20	60	10	1500		10/13 "
1	180	20	60	10	1500		10/13 "
1	180	15	60	10	1580		10/13 "
1	180	20	60	10	1450		10/13 "
1	180	20	60	10	1500		10/13 "
1	180	20	60	10	1750		10/13 "
1	280	30	50	10	1500		10/13 WEF 1018
1	300	30	10	10	1880	OK - changed to 1800	10/13 WEF 1018

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1977 JOURNEY			RECORD OF TIME					RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE MPG.	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. GASE CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
BROUGHT FORWARD							2540.5	NO.	LBS.					
Oct 17	QD - DN	T. Lumb	1730	1900	1.5		2540.5	2	300	30	5	50	1800	John L. 616-17941
" 18	DN - QD	"	1600	1730	1.5		2511.5	2	300	30	6	50	1800	"
" 23	QD - Morden	"	1600	1900	3.0		2514.5	3	450	30	6	50	1950	Oil Changed to W-65
" 24	Morden - QD	"	900	1230	3.5		2518.0	3	450	30	6	50	1950	"
Oct 31	QD - local	"	1000	1030	.5		2518.5	2	180	15	6	50	1580	"
Nov 15	"	"	1100	1200	1.0		2519.5	2	180	15	6	50	1580	"
1978														
Jan 17	QD - local	"	1100	1400	1.0		2520.5	2	300	30	6	50	1800	John L. 616-17941
Feb 2	"	"	1200	1300	1.0		2521.5	2	300	30	6	50	1800	"
Feb 13	QD - EE - Stuy - QD	"	1100	1230	1.5		2523.0	1	150	30	5	50	1600	"
Mar 12	QD - local	"	1200	1300	1.0		2524.0	2	300	30	6	50	1800	"
April 29	QD - Swan R.	"	1100	1215	1.25		2525.25	4	580	25	6	10	1900	QD Num. 39-2833, Ex. 34528
" 29	Swan R. - QD	"	1630	1715	25		2527	9	580	30	6	10	2000	Oil Changed to W-100
" 30	QD - Morden	"	1600	1900	3.0		2530	9	500	30	6	50	1980	"
" 30	QD - EE - W-65 - Min	"	1030	1100	1.0		2531	1	150	30	6	50	1800	"
May 1	Min - QD	"	1700	2030	3.5		2534.5	4	500	30	6	50	1980	"
June 27	QD - local	"	2100	2130	.5		2535.5	1	150	30	6	50	1800	"
" 30	"	"	2000	2015	.25		2535.75	1	150	30	6	50	1800	"
" 30	QD - BN	"	1600	1820	2.5		2537.75	4	500	30	6	50	1980	"
July 1	BN - Morden	"	1900	1945	.75		2538.50	4	500	30	6	50	1980	"
" 3	Morden - QD	"	1700	2030	3.5		2542.00	1	150	30	6	50	1800	"
" 31	QD - Morden	"	1900	1945	.75		2542.75	1	150	30	6	50	1800	"
" 23	Morden - QD	"	1900	2215	3.25		2546.00	1	150	30	6	75	1625	"
" 29	QD - Ft. McMurray	"	1200	1630	4.30		2550.30	2	300	30	6	200	1830	"
" 30	Ft. McMurray - QD	"	900	1300	4.00		2554.30	2	300	30	6	150	1830	"
Aug 2	Kelowna - Edmonton	"	800	1300	5.00		2559.30	4	350	30	6	150	1830	"
" 3	Edmonton - QD	"	1100	1545	4.5		2563.80	4	350	30	6	150	1830	"
" 4	QD - local	"	1500	1900	1.0		2564.80	4	350	30	6	150	1830	"
" 26	QD - Morden	"	1700	2000	3.0		2567.80	4	350	30	6	150	1830	"
" 31	"	"	1500	1500	0.0		2567.80							"

JOURNEY		CREW		RECORD OF TIME					RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE MPG.	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. GASE CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER	
BROUGHT FORWARD							2579.5	NO.	LBS.						
Nov 16/78	QD - local	T. Lumb	1215	1315	1.0		2580.5	2	300	30	6	110	1800	Oil changed W-65	T. Lumb
Jan 10/78	"	"	1600	1630	.5		2581.0	1	150	30	6	110	1500	"	T. Lumb
Mar 3/78	QD - Thompson	"	900	1300	4.0		2585.0	1	130	30	6	110	1570	"	T. Lumb
May 23/78	QD - Bala - QD	"	500	1000	5.0		2590.0	1	130	30	6	110	1570	Oil changed W-100	T. Lumb
May 27	QD - EE	"	800	1330	5.5		2595.5	4	350	30	6	150	1950	"	T. Lumb
June 3	EE - Ft. McMurray	"	1100	1300	2.0		2597.5	4	350	30	6	150	1950	"	T. Lumb
June 4	Ft. McMurray - Kelowna	"	600	945	3.5		2599.5	1	130	30	6	120	1800	"	T. Lumb
" 4	Kelowna - Regina - P.B.	"	1100	1330	2.5		2602.0	1	130	30	6	120	1800	"	T. Lumb
" 4	P.B. - QD	"	1900	1930	.5		2602.5	1	130	30	6	120	1800	"	T. Lumb
July 22	QD - local	"	1800	1830	.5		2603.0	2	265	30	6	50	1670	"	T. Lumb
Aug 2	QD - Regina	"	930	1200	2.5		2605.5	1	130	30	6	50	1570	"	T. Lumb
" 2	Regina - QD	"	1930	2230	3.5		2609.0	1	130	30	6	50	1570	"	T. Lumb
2 Oct 78	"	"	-	-	-		2609.0							"	T. Lumb
Nov 18	QD - local	T. Lumb	1300	1400	1.0		2610.0	1	130	30	6	50	1570	"	T. Lumb
1979															
April 6	QD - BN	T. Lumb	730	1030	3.5		2613.5	1	130	30	6	50	1570	"	T. Lumb
" 9	BN - QD	"	1930	1900	.25		2613.75	1	130	30	6	50	1570	"	T. Lumb
" 16	QD - Bala - QD	"	1030	1730	5.5		2619.25	2	265	30	6	50	1620	Oil changed W-100	T. Lumb
" 25	QD - Edmonton - QD	"	1300	1500	1.5		2620.75	2	265	30	6	50	1620	"	T. Lumb
May 15	QD - Thompson - QD	"	1930	1500	3.5		2624.25	1	130	30	6	50	1570	"	T. Lumb
June 9	QD - EE - Bala	"	1100	1430	3.5		2627.75	3	450	30	6	80	1950	"	T. Lumb
" 10	Bala - QD	"	1030	1200	1.5		2629.25	3	450	30	6	80	1950	"	T. Lumb
July 9	QD - BN	"	930	1100	1.5		2630.75	2	260	30	6	50	1600	"	T. Lumb
" 9	BN - Bala - Swan R.	"	1530	1530	1.5		2632.25	2	260	30	6	50	1600	"	T. Lumb
" 9	Swan R. - QD	"	1630	1730	1.0		2633.25	2	260	30	6	50	1600	"	T. Lumb
" 12	QD - local	"	1700	1730	.5		2633.75	1	120	20	6	30	1500	"	T. Lumb
" 17	QD - BN	"	800	1030	2.5		2636.25	1	120	20	6	30	1500	"	T. Lumb
" 17	BN - QD	"	2200	2430	2.5		2638.75	1	150	30	6	50	1630	"	T. Lumb
" 19	QD - Bala - QD	"	1700	1900	2.0		2640.75	1	120	20	6	50	1630	"	T. Lumb
" 24	QD - local	"	800	1230	3.5		2644.25	1	130	30	6	50	1630	"	T. Lumb
Aug 28	QD - Regina - QD	"	1730	1610	3.5		2647.75	2	265	30	6	50	1630	"	T. Lumb

JOURNEY			CREW	RECORD OF TIME					7. TOTAL AIR TIME SINCE MFG.
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME			
BROUGHT FORWARD								2649.0	
Sept 7							2649.0		
Sept 12	QD - WPG	T. Lumb			16.00	19.00	3.0	2652.0	
Sept 14	Wpg - QD	T. Lumb			17.00	15.00	2.0	2654.0	
Nov 20	QD - P. K. Bow. ut	"			17.00	18.00	1.0	2655.0	
Feb 17/80	QD - local	"			15.00	16.00	1.0	2656.0	
Feb 23/80	QD - Brandon	"			10.00	12.30	2.5	2658.5	
" 24 "	Brandon - QD	"			11.30	14.00	2.5	2661.0	
Feb 29	QD - DN	"			16.00	18.00	2.0	2663.0	
Mar 1	DN - Gladstone	"			12.00	12.30	.5	2663.5	
Mar 1	Gladstone - QD	"			15.30	17.30	2.0	2665.5	
" 20 "	QD - Thompson	"			9.30	13.00	3.5	2669.0	
April 3	QD - Thompson	"			16.30	18.00	1.5	2670.5	
" 3 "	Thompson - QD	"			23.00	1.00	2.0	2672.5	
" 9 "	QD - Thompson	"			9.00	10.30	1.5	2674.0	
" 9 "	Thompson - QD	"			16.00	17.30	1.5	2675.5	
" 10 "	QD - Luf R.	"			16.30	18.30	2.0	2677.5	
" 10 "	Luf R. - QD	"			22.30	24.00	1.5	2679.0	
MAY 6	QD - Luf R.	"			16.30	18.30	2.0	2681.0	
" 7 "	Luf R. - QD	"			8.30	10.30	2.0	2683.0	
" 19 "	QD - SWAN R.	"			7.30	9.30	2.0	2685.0	
" 19 "	SWAN R. - QD	"			11.30	12.30	1.0	2686.0	
June 23	QD - Morden	"			12.00	15.00	3.0	2689.0	
TOTAL THIS PAGE								2716.0	

RECORD OF WEIGHTS						REMARKS	SIGNATURES								
8.	TOTAL PERSONS ON BOARD	9.	FUEL IMP. GAL.	10.	OIL IMP. GAL.			11.	WEIGHT EQUIP. BAG-GAGE CARGO	12.	TOTAL WEIGHT AT T/O	13.	DEFECTS & AIRWORTHINESS CERTIFICATION	14.	PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER
NO.	LBS.														
1	110	30	6.0	110	I certify I have flown this aircraft in compliance with the provisions and conditions of inspection procedure prescribed in the Engineering and Inspection Manual and it is satisfactory.			T. Lumb							
1	130	30	6.0	110	OIL CHANGED TO W 55			T. Lumb							
2	125	20	6.0	110				T. Lumb							
2	300	30	6.0	150				T. Lumb							
2	300	30	6	150				T. Lumb							
2	300	30	6	150				T. Lumb							
2	300	30	6	150				T. Lumb							
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2	300	30	6	150		T. Lumb									
2	300	30	6	150											

JOURNEY			CREW		RECORD OF TIME				7. TOTAL AIR TIME SINCE MFG.
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	FLIGHT TIME			
BROUGHT FORWARD								2746.6	
Aug 24	MORDEN LOCAL	T. LUMB	1430	1530	1.0	1.0	2746.6		
Nov 10	"	"	1350	1420	1.0	1.0	2747.6		
22 June 86	CCT	E & F A Renewal - EXPIRED					2747.6		
29 June 86	MORDEN LOCAL	E. HELPS	1100	1120		0.3	2747.9		
29 June 86	I certify that all applicable requirements of the E & F annual have been met and the aircraft is airworthy.								
25 Aug 86	MORDEN LOCAL	E. HELPS	1915	1935		0.3	2748.2		
Sept 10/86	"	T. LUMB	1520	1550		0.5	2748.7		
Oct 20/86	MORDEN - STONEWELL	T. LUMB	710	1010		1.0	2749.7		
" 22/86	STONEWELL - LOCAL	"	1630	1700		0.5	2750.2		
July 27/87	code for 2 of A renewal (expired)						2750.2		
July 27/87	STONEWELL	T. LUMB				0.2	2750.4		
July 27/87	I certify that all applicable requirements of the E & F annual have been met and the aircraft is airworthy.								
Aug 15/87	STONEWELL - WYKE MT.	T. LUMB	1900	1930	0.5	0.5	2750.9		
Aug 26	STONEWELL - ST. FRANCIS	T. LUMB	1800	2000	1.0	1.0	2751.9		
Sept 11	STONEWELL - WYKE MT.	"	1700	1740	0.6	0.6	2752.5		
Oct 7	STONEWELL LOCAL	"	1730	1800	0.5	0.5	2753.		
Oct 16	"	"	1800	1820	0.5	0.5	2753.5		
April 3/88	STONEWELL LOCAL	"	1600	1700	1.0	1.0	2754.5		
May 10/88	STONEWELL - WYKE MT.	"	1200	1300	1.0	1.0	2755.5		
9 Aug 88	CCT						2755.5		
9 Aug 88	STONEWELL LOCAL	E. HELPS				0.2	2755.7		
9 Aug 88	I certify that all applicable requirements of the E & F annual have been met and the aircraft is airworthy.								
Oct 4/88	STONEWELL LOCAL	T. LUMB	1630	1730	1.0	1.0	2756.7		
April 3/89	"	"	1210	1240	0.5	0.5	2757.2		
May 16/89	"	"	1700	1730	0.5	0.5	2757.7		
4 Aug 89	ANNUAL AIRWORTHINESS INFO REPORT SUBMITTED UNDER PART 21 CAR 37(1)(b)								
TOTAL THIS PAGE								2757.7	

RECORD WEIGHTS												REMARKS	SIGNATURES
8.	TOTAL PERSONS ON BOARD	9.	FUEL IMP. GAL.	10.	OIL IMP. GAL.	11.	WEIGHT EQUIP. BAGG. GASES CARGO	12.	TOTAL WEIGHT AT T/O	13.	DEFECTS & AIRWORTHINESS CERTIFICATION	14.	PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
NO.	LBS.												
2	220	22	6	10	1500								
2	220	18	6	10	1480								
I certify I have inspected this aircraft in compliance with the Condition and Conformity Inspection procedure prescribed in the Engineering and Inspection Manual and it is airworthy.													
I hereby certify I have flown this aircraft and its performance, handling, fuel consumption, oil quantity, engine, gear and landing gear, etc. were equivalent to the standard of a													
airworthy													
I have not determined to be airworthy													
2	320	22	1	10	1480								
2	300	20	6	10	1530								
1	180	28	6	20	1440								
2	200	15	6	20	1490								
I hereby certify I have completed the nearest equivalent to a 100 hr. periodic inspection of the aircraft as prescribed in the Condition and Conformity Inspection procedure prescribed in the Engineering and Inspection Manual and it is airworthy.													
I hereby certify I have flown this aircraft and its performance, handling, fuel consumption, oil quantity, engine, gear and landing gear, etc. were equivalent to the standard of a													
airworthy													
I have not determined to be airworthy													
2	200	30	6	20	1490								
3	350	30	6	20	1600								
3	350	25	6	20	1600								
2	225	20	6	20	1400								
2	225	25	6	20	1380								
2	225	25	6	20	1490								
2	225	20	6	20	1400								
I certify I have inspected this aircraft in compliance with the Condition and Conformity Inspection procedure prescribed in the Engineering and Inspection Manual and it is airworthy.													
I hereby certify I have flown this aircraft and its performance, handling, fuel consumption, oil quantity, engine, gear and landing gear, etc. were equivalent to the standard of a													
airworthy													
I have not determined to be airworthy													
1	225	16	6	20	1380								
1	225	16	6	20	1330								
1	225	25	6	20	1430								
I certify I have inspected this aircraft in compliance with the Condition and Conformity Inspection procedure prescribed in the Engineering and Inspection Manual and it is airworthy.													

JOURNEY		CREW		RECORD OF TIME				7. TOTAL AIR TIME SINCE MFG.	RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	FLIGHT TIME	8. TOTAL PERSONS ON BOARD		9. FUEL IMP. GAL.	10. OIL IMP. GAL.	11. WEIGHT EQUIP. BAGG. GASES CARGO	12. TOTAL WEIGHT AT T/O	13. DEFECTS & AIRWORTHINESS CERTIFICATION	14. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER	
BROUGHT FORWARD									NO.	LBS.					
Aug 21/89	MORDEN	T. LUMB	1700	1800	1.0	1.0	2758.2	1	120	20	6	20	1400		T. Lumb
Aug 22	" LOCAL	"	1800	1900	1.0	1.0	2759.2	3	260	30	6	20	1580		T. Lumb
23	STONEWELL	"	1900	2000	1.0	1.0	2760.2	3	240	28	6	20	1560		T. Lumb
Oct 21	" LOCAL	"	1700	1800	1.0	1.0	2761.2	2	200	16	6	20	1550		T. Lumb
" 27 1990	"	"	1730	1830	1.0	1.0	2762.2	3	350	12	6	20	1680		T. Lumb
May 15	STONEWELL local	T. LUMB	1800	1830	0.5	0.5	2763.2	2	225	20	6	20	1590		T. Lumb
May 30	"	"	1900	2000	1.0	1.0	2764.2	2	230	16	6	20	1520		T. Lumb
June 30	MORDEN	"	1700	1745	0.75	0.75	2765.2	1	150	28	6	20	1500		T. Lumb
July 16	STONEWELL	"	1900	1945	0.70	0.70	2766.7	1	150	30	6	20	1550		T. Lumb
July 18	" LOCAL	"	2000	2030	0.30	0.30	2767.7	2	300	20	6	20	1630		T. Lumb
July 30	St. Francis	"	2000	2015	0.50	0.50	2768.2	1	150	20	6	20	1630		T. Lumb
TOTAL THIS PAGE									Same correction 1 hr. less Jan 30 entry						
TOTAL THIS PAGE									36.74						
Aug 10/90	LYNCREST	W. BIRCH	19:50	20:30	0.5	0.5	2767.0	1	180	30	1.5	10	1561		W. Birch
SEPT 2	ZHODA	"	9:45	10:27	0.7	0.9	2767.7	1	180	28	1.5	10	1518		W. Birch
" 2	LYACREST	"	11:08	11:52	0.7	0.9	2768.6	1	180	20	1.5	10	1475		W. Birch
" 3	LYNCREST	K. Copp	19:45	20:43	1.0	1.3	2669.4	2	320	30	1.5	10	1701		K. Copp
SEPT 3	LYNCREST	K. Copp	11:29	11:57	0.5	0.6	2669.9	3	440	33	1.5	10	1761		K. Copp
" 3	LYNCREST	K. Copp	14:21	14:40	0.3	0.5	2670.2	3	440	15	1.5	10	1705		K. Copp
" 5	"	W. BIRCH	19:10	20:12	1.0	1.2	2671.2	1	180	30	1.5	10	1561		W. Birch
" 7	"	K. Copp	14:28	20:41	0.9	1.0	2672.2	3	530	23	1.5	10	1861		K. Copp
" 8	"	W. BIRCH	14:43	20:06	0.4	0.8	2673.6	3	440	20	1.5	10	1823		W. Birch
" 10	"	W. BIRCH	18:41	19:39	1.0	1.3	2673.6	3	540	25	1.5	10	1515		W. Birch
" 14	"	K. Copp	19:34	19:57	0.5	0.6	2674.1	2	140	30	1.5	10	1581		K. Copp
" 16	Barnes	K. Copp	8:58	9:22	0.5	0.6	2674.7	4	460	26	1.5	10	2000		K. Copp
TOTAL THIS PAGE									36.74						

I hereby certify that I have completed the aircraft inspection in accordance with the provisions of the Engineering and Inspection Manual and it is airworthy.

JOE SKRIVINSKY, DAT 10/10/90

WGM05181

Aug 10/90	LYNCREST	W. BIECH	1940	2020	0.5	0.5	2769.0		1	180	30	10	10	1561	
Sept 2	ZHODA	"	1945	2027	0.7	0.7	2769.7		1	180	28	10	10	1518	
" 2	LYNCREST	"	1908	1952	0.7	0.7	2769.4		1	180	20	10	10	1475	
" 3	LYNCREST	K. Copp	1945	2042	1.0	1.3	2769.4		2	920	30	15	10	1701	
Sept 8	LYNCREST	K. Copp	1129	1157	0.5	0.6	2769.9		3	440	33	15	10	1761	
" 3	LYNCREST	K. Copp	1941	1942	0.3	0.5	2770.2		3	440	15	15	10	1705	
" 5	"	W. BIECH	1930	2012	1.0	1.2	2771.2		1	180	30	10	10	1561	
" 7	"	K. Copp	1928	2004	0.7	1.0	2772.2		3	530	23	15	10	1861	
" 8	"	W. BIECH	1949	2006	0.4	0.8	2772.6		3	440	20	15	10	1823	
" 10	"	W. BIECH	1921	1939	1.0	1.2	2773.6		3	540	25	15	10	1895	
" 14	"	K. Copp	1934	1951	0.5	0.6	2774.1		1	140	30	15	10	1581	
" 16	Barnes	K. Copp	1958	1952	0.5	0.4	2774.7		4	660	26	10	10	2008	
TOTAL THIS PAGE															

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[illegible]

Aircraft C-FZNR has been transferred to the Owner Maintenance category and complies with the requirements of CAR 507.03(6)

Date: JULY 21-00 Signature: W. Brish
Lic. No. PA 36082

Aircraft C-FZNR shall be maintained in accordance with Standard 625 Appendix B & C

Date: JULY 21-00 Signature: W. B. Buel
Lic. No: PA 36082

10. Fish 36002
10. Fish -
10. Fish

Aircraft C-FZNR has been transferred to the Owner Maintenance category and complies with the requirements of CAR 507.03(6)

Date: JULY 21-00 Signature: W. Briel
Lic. No: PA 36082

Aircraft C-FZNR shall be maintained in accordance with Standard 625 Appendix B & C

Date: JULY 21-00 Signature: W. Briel
Lic. No: PA 36082

Aircraft C-FZNR has been transferred to the Owner Maintenance category and complies with the requirements of CAR 507.03(6)

Date: JULY 21-00 Signature: W. Brich
Lic. No: PA 36082

Aircraft C-FZNR shall be maintained in accordance with Standard 625
Appendix B & C

Appendix B & C

Date: JULY 21-00 Signature: W. Birch
Lic. No: PA 36082

	W. Buch
2	W. Buch

JOURNEY		CREW		RECORD OF TIME			RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. TOTAL AIR TIME SINCE MPG	8. TOTAL PERSONS ON BOARD	9. FUEL IMP. GAL.	10. OIL IMP. GAL.	11. WEIGHT EQUIP. BAG. GAGE CARGO	12. TOTAL WEIGHT AT T/O	13. DEFECTS & AIRWORTHINESS CERTIFICATION	14. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
2000													
						BROUGHT FORWARD							
AUG 22	YAV. EYERHART-YAV	W. BIRCH			4.7	2922.2							
OCT 3	YAV. LOCAL	"			.7	2926.6							
" 6	YAV. LOCAL	"			.6	2927.2							
" 8	YAV. SHAL L. YAV	"			4.2	2931.4							
" 11	YAV. AURORA - YAV	"			1.8	2933.2				REPLACED OIL PRESS. TEMP. WITH WESTACH G.M.D.			W. Fitch
" 12	YAV. STONBACH - YAV	"			1.2	2934.4							
" 15	YAV. LYNDSEY - YAV	K. Copp			.8	2935.2							
" 15	YAV. MAGGROUR - YAV	W. BIRCH			1.7	2936.9							
NOV 3	YAV. LYNDSEY - YAV	"			1.0	2937.9							
" 4	YAV. LYNDSEY - YAV	"			1.2	2939.1							
" 24	YAV. GIMLI - YAV	"			1.3	2940.4							
" 25	"	"			1.3	2941.7							
" 27	"	"			1.4	2943.1							
JAN. 10	YAV. LOCAL	"			.8	2943.9							
FEB. 3	YAV. "	"			.2	2944.1							
APR 30	YAV. GIMLI - YAV	"			1.1	2945.2							
MAY 12	YAV. LYNDSEY - YAV	"			.7	2945.9							
MAY 12	"	"			.8	2946.7							
MAY 26	YAV. GIMLI - YAV	K. Copp			3.2	2949.9							
JUNE 2	YAV. GIMLI - YAV	W. BIRCH			1.0	3000.9							
" 3	YAV. SHAL L. - YAV	"			3.4	3004.3							
" 8	YAV.	"			.8	3005.1							
" 9	YAV. STONBACH - YAV	"			1.0	3006.1							
" 10	YAV. YBR - YAV	"			3.0	3009.1							
" 16	YAV. YBR - YAV	"			.9	3010.0							
" 17	YAV. KQG - YON - YAV	"			3.5	3013.5							
" 17	YAV.	"			1.0	3014.5							
JULY 1	YAV. YAV - YAV	"			1.7	3016.2							W. Fitch
AUG 3	ANNUAL (100HR) INSPECTION					3016.2							
AUG 7	TEST FLIGHT				.5	3016.7							
AUG 8					1.1	3017.8							
AUG 12					2.0	3019.8							
AUG 12					1.1	3020.9							

JOURNEY		CREW		RECORD OF TIME			RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. TOTAL AIR TIME SINCE MPG	8. TOTAL PERSONS ON BOARD	9. FUEL IMP. GAL.	10. OIL IMP. GAL.	11. WEIGHT EQUIP. BAG. GAGE CARGO	12. TOTAL WEIGHT AT T/O	13. DEFECTS & AIRWORTHINESS CERTIFICATION	14. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
2001													
						BROUGHT FORWARD							
AUG 16	YAV. YON - YAV	K. Copp			1.0	3020.9							
" 16	"	W. BIRCH			2.0	3022.9							
" 19	"	"			.8	3023.7							
" 22	"	"			1.1	3024.8							
" 26	"	"			1.9	3026.7							
" 27	"	"			1.7	3028.4							
" 30	"	"			1.0	3030.4							
" 31	"	"			1.7	3032.1							
SEPT 3	"	"			2.2	3034.3							
" 4	"	"			1.2	3035.5							
" 9	"	"			.7	3036.2							
" 14	"	"			6.5	3042.7							
" 15	"	"			1.0	3043.7							
" 16	"	"			4.5	3048.2							
" 20	"	"			.8	3049.0							
" 23	"	"			2.4	3051.4							
" 30	"	"			2.4	3053.8							
OCT 4	"	"				3054.4							
" 12	"	"			2.3	3056.7							
" 13	"	"			1.1	3057.8							
" 14	"	"			1.6	3059.4							
" 14	"	"			1.6	3061.0							
" 20	"	"			1.9	3062.9							
" 26	"	"			.9	3063.8							
" 29	"	"			1.3	3065.1							
NOV 3	"	"			1.1	3066.2							
" 4	"	"			1.9	3068.1							
" 5	"	"			1.2	3069.3							
" 6	"	"			.8	3069.1							
" 21	"	"			1.0	3070.1							
" 11	"	"			2.8	3072.9							

SEPT 23/01

FULL THROTTLE
LEVEL FLIGHTRPM - 2575
IAS - 125OAT - 0°C
ALT. - 2200

REPLACED BUSHINGS & BOLTS IN NOSE GEAR TOWING ARM

W. Fitch

21

JOURNEY		CREW		RECORD OF TIME			
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE DEPARTURE
2001							
BROUGHT FORWARD							3075.4
NOV. 15		W. BIRCH				1.3	3076.7
" 16		"				1.0	3077.7
" 20		"				2.0	3079.7
APR 11							
APR. 12		"				1.7	3080.4
" 13		"				1.0	3081.4
" 20		"				1.8	3082.2
" 22		"					
MAY 13		"				1.0	3083.2
" 18		"				1.7	3083.9
" 18		"				1.0	3084.9
" 19		"				1.2	3086.1
" 20		"				1.0	3087.1
" 2		"				1.8	3087.9
" 8		"				1.0	3088.9
" 13		"				1.4	3090.3
" 14		"				1.9	3092.2
" 15		"				1.6	3093.8
" 30		"				1.0	3094.8
JUNE 13		"				1.9	3096.7
" 16		"				1.6	3098.3
" 30		"				2.4	3100.7
JULY 1		"				1.6	3101.3
" 2		"					3101.3
" 6		"				1.6	3102.9
" 10		"				3.1	3106.0
" 4		"				1.7	3106.7
" 16		"				1.0	3107.7
" 20		"				1.5	3109.2
" 23		"				2.3	3111.5
" 24		"					
TOTAL THIS PAGE							

RECORD OF WEIGHTS						REMARKS		SIGNATURES					
9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. GAGE CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER							
NO.	LBS.												
2 MAIN WHEELS - REPLACED IN NEW WHEELS AND BRAKES WITH DISC BRAKES (USED PARTS) 2/NEW TIRES													
REMOVED HYDRA-SURFS - CHECKED, CLEANED & PAINTED REPLACED SHOCK RODS (1080HD)													
PARK BRAKE - REPLACED SEALS													
JULY 8 - 2002													
COMPRESSION TEST.													
#1 80/70													
#2 80/75													
#3 80/78													
#4 80/78													
REPLACED OIL SCREEN WITH ADAPTER & FILTER													

2 MAIN WHEELS - REPLACED INNER WHEELS AND BRAKES WITH DISC BRAKES (USED PARTS) 2/NEW TIRES

W. Birch

REMOVED HYDRA-SURFS - CHECKED, CLEANED & PAINTED REPLACED SHOCK RODS (J0804D)

W. Birch

PARK BRAKE - REPLACED SEALS

W. Birch

JULY 8 - 2002
COMPRESSION TEST.

#1 80/70

#2 80/75

#3 80/78

#4 80/78

REPLACED OIL SCREEN WITH ADAPTER & FILTER

W. Birch

Annual Inspection was carried out in accordance with STD 625 App B & C

W. Birch
31682

REMOVED ELT NARCO 16 INSTALLED ELT AR4490

K. Copp 316087

W. Birch

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME IN HRS. MINS.	9. TOTAL PERSONS ON BOARD	10. FUEL WGT. GAL.	11. OIL WGT. GAL.	12. WEIGHT EQUIP. BAGG. GAGE CARGO	13. TOTAL WEIGHT AT 7/8	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
2004														
Nov 26		W. BIRCH					3266.7							
DEC 13		"					3274.4							
NOV 19		"					3270.0							
FEB 6		"					3274.6							
" 11		"					3271.6							
" 12		"					3272.7							
MAR 26		"					3273.8							
APR 23		"					3274.9							
MAY 16		"					3279.3							
" 23		"					3277.8							
JUNE 6		"					3278.7							
" 11		"					3280.7							
" 26		"					3282.0							
JULY 8		"					3284.5							
" 16		"					3286.4							
" 28		"					3286.9							
AUG 3		"					3287.3							
" 6		"					3288.1							
" 16		"					3288.0							
" 19		"					3289.1							
" 20		"					3291.0							
" 28		"					3293.5							
SEP 3		"					3294.6							
" 5		"					3295.0							
" 7		"					3296.0							
" 8		"					3296.6							
" 11		"					3297.3							
" 18		"					3296.9							
" 31		"					3299.9							
" 32		"					3301.0							
" 34		"					3301.6							
" 39		"					3302.8							
TOTAL THIS PAGE														

Annual Inspection was carried out in accordance with STD 625 App B & C

W. Birch
FA 36082

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME IN HRS. MINS.	9. TOTAL PERSONS ON BOARD	10. FUEL WGT. GAL.	11. OIL WGT. GAL.	12. WEIGHT EQUIP. BAGG. GAGE CARGO	13. TOTAL WEIGHT AT 7/8	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENCE NUMBER
2006														
SEPT 20		W. BIRCH					3302.1							
Oct 9		"					3306.1							
" 10		"					3307.3							
" 13		"					3308.2							
" 16		"					3310.2							
" 22		"					3311.3							
" 24		C. KOTT					3312.6							
" 27		"					3313.8							
" 29		"					3315.3							
Nov 3		W. BIRCH					3316.3							
" 6		C. KOTT					3317.2							
" 11		K. LOPP					3318.4							
" 12		W. BIRCH					3318.4							
" 20		"					3319.9							
" 24		"					3320.6							
" 25		"					3321.6							
" 26		C. KOTT					3323.7							
" 27		"					3327.9							
Dec 10		W. BIRCH					3329.1							
Dec 1		W. BIRCH					3330.0							
JAN 18 2006		C. KOTT					3331.5							
" 28		"					3331.8							
FEB 3		"					3332.4							
" 4		"					3335.5							
" 11		W. BIRCH					3336.0							
" 12		"					3337.0							
" 23		"					3339.0							
" 25		C. KOTT					3341.3							
MAR 2		W. BIRCH					3342.2							
" 13		C. KOTT					3344.9							
APR 6														
" 6														
TOTAL THIS PAGE														

REPLACED STARTER WITH LAJALIC STARTER
REPLACED GENERATOR WITH ALTERNATOR

W. Birch
FA 36082

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. BAGG. CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER
BROUGHT FORWARD							3341.9							
APR 8		W. BIRCH				1.9	3341.9							
JULY 15		C. NEWICK				1.4	3343.3							
SEPT 22						2.1	3345.4							
MAY 10		W. BIRCH				1.2	3346.6							
JUN 4		"				.8	3347.4							
JUN 13		"				.4	3347.8							
JUN 14		"				1.1	3348.9							
JUN 21		"				2.1	3351.0							
JULY 2		"				.8	3351.8							
" 5		"				1.0	3352.8							
" 6		"				2.0	3354.8							
" 24		"				1.0	3355.8							
" 27		"				1.4	3357.2							
" 28		"				2.0	3359.2							
AUG 3		"				1.0	3360.2							
" 4		"				1.2	3361.4							
" 7		"				.8	3362.2							
" 9		"				2.0	3364.2							
" 12		"				1.9	3366.1							
" 17		"				1.9	3368.0							
" 21		C. NETT				.9	3368.9							
" 21		W. BIRCH				.5	3369.4							
" 24		"				1.7	3371.1							
" 25		"				2.0	3373.1							
" 28		K. COPP				.6	3373.7							
" 29		W. BIRCH				1.6	3375.3							
" 30		"				1.2	3376.5							
" 31		"				2.8	3379.3							
SEPT 2		"				1.0	3380.3							
" 2		K. COPP				1.1	3381.4							
" 2		W. BIRCH				.4	3381.8							
TOTAL THIS PAGE														

Annual Inspection was carried out in accordance with STD 624 App B & C
Flight Test — Performance conforms to type

AUTHORISED CERTIFIED BY EX-107

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. BAGG. CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER
BROUGHT FORWARD							3382.6							
SEPT 14		W. BIRCH				1.9	3384.5							
" 16		"				1.6	3386.1							
" 17		"				.3	3386.4							
Oct 1						.5	3386.9							
" 3		W. BIRCH				.5	3387.4							
NOV 10		"												
NOV 11		"												
NOV 12		W. BIRCH												
NOV 12		W. BIRCH				1.3	3388.7							
DEC 15		"				1.0	3389.7							
" 17		"				.5	3390.2							
" 20		"				1.0	3391.2							
JAN 2		"				.9	3392.1							
JAN 3		"				1.0	3393.1							
" 6		"				1.5	3394.6							
" 20		"				1.0	3395.6							
FEB 22		"				1.1	3396.7							
" 23		"				1.5	3400.2							
" 29		"				1.3	3401.5							
" 30		K. COPP				1.4	3402.9							
MAR 10		K. COPP W. BIRCH				2.1	3405.0							
" 15		W. BIRCH				1.4	3406.4							
" 18		"				1.0	3407.4							
" 23		"				1.4	3408.8							
APR 5		"				1.0	3410.8							
MAY 24		"				.5	3411.3							
" 31		"				3.0	3414.3							
JUNE 14		"				.0	3414.3							
" 19		"				1.0	3415.3							
" 20		P. Charles Rosario				1.0	3416.3							
" 20		W. BIRCH				.9	3417.2							
" 25														
TOTAL THIS PAGE														

Annual Inspection was carried out in accordance with STD 624 App B & C
Flight Test — Performance conforms to type

REPAIRED RUBBER ENGINE MOUNTS & BOLTS WITH NEW.
REPLACED MUFFLER WITH NEW.

REPLACED MASTER BRACE CYL WITH STEEL'S A/C
MODIFIED SAB-01 MASTER CYL WITH VENTED RESERVOIR

29

JOURNEY		CREW		RECORD OF TIME			RECORD OF WEIGHTS					REMARKS		SIGNATURES	
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. CASE CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER	
BROUGHT FORWARD															
JUN 29		W. BIRCH				1.1	3421.7							W. Birch	
JUL 24		"				1.2	3420.8								
" 26		"				1.6	3420.6								
AUG 5		"				1.7	3421.3								
" 6		"				1.4	3422.7								
" 10		"				1.9	3422.4								
" 15		"				1.9	3422.5								
" 16		"				1.2	3426.7								
" 19		"				1.8	3426.7								
" 24		"				2.0	3424.8								
" 27		"				1.0	3424.6								
" 28		"				1.7	3431.2								
" 29		"				1.9	3432.1								
SEPT 3		"				1.6	3429.7								
" 4		"				1.3	3428.0								
" 10		"				1.0	3435.0								
" 11		"				1.6	3436.6								
" 19		K. COPP				1.1	3436.7							W. Birch	
" 20		W. BIRCH				1.0	3437.7							W. Birch	
OCT 19		W. BIRCH				1.2	3432.9							W. Birch	
OCT 20		"				1.9	3434.8							W. Birch	
" 26		"				1.1	3440.9								
" 29		"				2.0	3443.1								
" 31		"				1.2	3442.3								
NOV 1		"				1.0	3445.3								
" 23		"				1.9	3446.2								
" 26		"				1.2	3447.4								
APR 10 2009		"				1.2	3447.4								
MAY 22		"				1.2	3447.4								
MAY 23		"				3.0	3450.4								
MAY 29		"				1.0	3451.6								
MAY 31		"				1.9	3452.5								
JUN 13		"				2.1	3454.6								
TOTAL THIS PAGE															

Annual Inspection was carried out in accordance with STD 624 App B & C
Flight Test - Performance conforms to type

30

JOURNEY		CREW		RECORD OF TIME			RECORD OF WEIGHTS					REMARKS		SIGNATURES	
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAG. CASE CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER	
BROUGHT FORWARD															
JUN 21		W. BIRCH				2.0	3454.6							W. Birch	
JUL 1		"				1.7	3457.3							PA 34582	
" 3		"				1.7	3459.0								
AUG 29		"				1.3	3460.5								
SEPT 4		"				1.3	3461.6								
" 5		"				1.0	3462.6								
" 19		"				2.1	3464.7								
" 26		K. COPP				1.8	3465.5							K. COPP	
OCT 10		W. BIRCH				1.4	3466.9							W. Birch	
" 18		K. COPP				1.6	3468.5							K. COPP	
" 19		W. BIRCH				1.6	3467.1							W. Birch	
" 22		"				1.6	3467.7								
" 24		C. RUFF				1.1	3468.8							C. Ruff	
NOV 03		C. RUFF				1.1	3469.9							C. Ruff	
" 11		K. COPP				1.9	3477.8							K. COPP	
NOV 13		W. BIRCH				2.2	3474.0							W. Birch	
" 14		"													
DEC 5		W. BIRCH				1.1	3475.1							W. Birch	
" 21		"				1.1	3476.2								
JAN 14-2010		"				1.7	3477.9								
" 17		"				1.0	3478.9								
FEB 27		"				1.1	3480.2								
MAR 22		"				1.1	3481.1								
APR 11		"				1.6	3481.7								
" 17		"				1.0	3482.7								
" 18		"				1.0	3483.1								
" 22		"				1.2	3484.3								
" 24		K. COPP				1.0	3485.3							K. COPP	
" 25		W. BIRCH				1.0	3486.3							W. Birch	
MAY 6		"				1.0	3486.3								
SEP 29		"				1.1	3487.4								
OCT 1		W. BIRCH				1.1	3487.4							W. Birch	
TOTAL THIS PAGE															

Annual Inspection was carried out in accordance with STD 624 App B & C
Flight Test - Performance conforms to type

REMOVED COMH NARCO-11 - INSTALLED COMH Test-11

ENGINE REINSTALLED - O.H. "O" TIT BY UNIVERSAL AERO - ENGINE REINSTALLED

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE MFG	9. TOTAL PERSONS ON BOARD	10. FUEL HP. GAL.	11. OIL HP. GAL.	12. WEIGHT EQUIP. BAG. BAGG. CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER
2010														
BROUGHT FORWARD														
Aug 3		W. BIRCH			1.8	3529.2								N. Copp
" 8		"			2.0	3531.2								
" 8		"			1.4	3532.6								
" 10		"			2.0	3534.6								
" 11		"			2.8	3537.4								
" 16		A. Copp			1.6	3539.0								A. Copp
" 21		W. BIRCH			1.5	3540.5								W. Birch
" 23		"			1.5	3542.0								
NOV 1		"			1.1	3543.1								
" 2		"			1.0	3544.2								
" 20														
" 20		W. BIRCH			1.8	3546.0								W. Birch
" 20		"			1.7	3547.7								
2011 MAR 26		"			1.1	3548.8								
MAR 28		"			1.0	3549.8								
APR 2		"			1.1	3550.9								
" 7		"			1.0	3551.9								
" 11		"			1.1	3553.0								
" 20		"			1.2	3554.2								
" 27		"			1.0	3555.2								
" 28		"			1.0	3556.2								
MAY 6		"			1.0	3557.2								
" 14		"			1.0	3558.2								
JUN 5		"			1.0	3559.2								
" 10		"			1.0	3560.2								
" 11		"			1.0	3561.2								
" 12		A. Copp			2.3	3563.5								A. Copp
" 16		W. BIRCH			1.4	3564.9								W. Birch
" 16		"			1.7	3566.6								
AUG 24		W. BIRCH			1.0	3567.6								W. Birch
SEP 9		A. Copp			1.6	3569.2								A. Copp
Sept 10		W. BIRCH			1.4	3570.6								W. Birch
Sept 10		A. Copp			1.4	3572.0								A. Copp
2011														

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME SINCE MFG	9. TOTAL PERSONS ON BOARD	10. FUEL HP. GAL.	11. OIL HP. GAL.	12. WEIGHT EQUIP. BAG. BAGG. CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS & AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MAKING ENTRY AND LICENSE NUMBER
2011														
BROUGHT FORWARD														
Sept 14		KAROLINA VIKO			1.2	3523.6								K. Copp
Oct 1		"			2.7	3526.3								
Oct 6		KAROLINA VIKO			1.1	3527.4								
Oct 9		Wojciech Koch			2.6	3529.9								
Oct 23		KAROLINA VIKO			1.3	3531.2								
Oct 24		Rena Prater			1.3	3532.5								
Oct 24		Wojciech Koch			1.3	3533.8								
Nov 5		Rena Prater			1.3	3535.1								
Nov 11		KAROLINA VIKO			1.3	3536.4								
Dec 4		Rena Prater			1.3	3537.7								
Dec 17		Wojciech Koch			1.6	3539.3								
2012														
Feb 4		Wojciech Koch			1.6	3540.9								
Feb 18		Wojciech Koch			1.1	3542.0								
Mar 12		KAROLINA VIKO			1.8	3543.8								
Mar 17/12		Wojciech Koch			1.5	3545.3								
Apr 03/12		A. RIFAI			1.3	3546.6								
May 26/12		A. Copp			2.8	3549.4								
Jun 2/12		W. Koch			2.3	3551.7								
Jun 6/12		W. Koch			1.5	3553.2								
Nov 7/12						3554.7								
Nov 8		K. COPP			1.4	3556.1								
Nov 22/12		Rena Prater			1.3	3557.4								
2013														
2013/13		A. Copp			1.8	3559.2								
Oct 7/2015						3561.2								
Nov 20/2017		A. Copp (Rena)			1.2	3562.4								
Dec 3/17	YAV Local	A. Lead/A. Copp	1834	1625	0.6	10	3563.4							
Jan 24/18	YAV Local	A. Lead/A. Copp	1730	1554	0.6	0.9	3564.3							
July 12/18	YAV - 3000	A. Lead/A. Copp	2002	2032	0.5	0.7	3564.8							
TOTAL THIS PAGE														

JOURNEY		CREW		RECORD OF TIME				RECORD OF WEIGHTS					REMARKS	SIGNATURES
1. DATE	2. FROM PREVIOUS LINE TO	3. NAMES	4. UP	5. DOWN	6. AIR TIME	7. FLIGHT TIME	8. TOTAL AIR TIME HOURS	9. TOTAL PERSONS ON BOARD	10. FUEL IMP. GAL.	11. OIL IMP. GAL.	12. WEIGHT EQUIP. BAGS, GAGE, CARGO	13. TOTAL WEIGHT AT T/O	14. DEFECTS AIRWORTHINESS CERTIFICATION	15. PILOT OR ENGINEER MARINE ENTRY AND LICENCE NUMBER
BROUGHT FORWARD →								NO.	LBS.					
23 Jul 2013	On Water - CYG	R Leach / R Lipp	1845	1908	0.4	0.6	3564.4							R Lipp 6211087
28 Sept 2013							3564.4						Annual inspection was carried out in accordance with STD 624 APP B/C	R Lipp 6211087
4 Oct 13	YAV Level	R Leach / R Lipp	0927	1004	0.6	0.9	3565.0							R Lipp 6211087
25 Oct 19							3565.0						Annual inspection was carried out in accordance with STD 624 APP B/C	R Lipp 6211087



Transport Canada Transports Canada

ANNUAL AIRWORTHINESS INFORMATION REPORT

COPP, KELLY L.

92 ELM DRIVE

OAKBANK, MANITOBA, CANADA

FORM 1.79

COMPLETE FORM AND SUBMIT NO LATER THAN THE DUE DATE

See reverse for change of address and form instructions.

Aircraft operated pursuant to CAR IV or CAR VII (inspection information not required) <input type="checkbox"/>		Issued (yyyy-mm-dd) 2021-03-26	Due (yyyy-mm-dd) 2021-03-30
Date of the most recent annual or 100 hour inspection (yyyy-mm-dd) 2019-10-25		Registration Mark FZNR	Type Certificate number NOT CERTIFICATED
AMO, AME or owner who/which conducted & certified this inspection Name COPP, KELLY L. <input type="radio"/> AMO Number <input type="radio"/> AME <input checked="" type="radio"/> Owner		Type of Flight Authority OWNER MAINT	Type of Registration PRIVATE
Has the aircraft been damaged since last report? <input type="radio"/> Yes (if yes, enter date) <input checked="" type="radio"/> No Date of repair certification (yyyy-mm-dd)		AIRCRAFT BASE OF OPERATION Country CANADA Province/State MANITOBA Municipality OAKBANK Airport ST. ANDREWS Other (House/Arm/etc) 24 TC Region PNR	
Reporting year 2020		Total hours flown since new – to Dec 31 of the reporting year 3565 hr	
Optional for air operators & flight training units Specialty Hours Training 0 hr		Hours flown – Jan 01 to Dec 31 of the reporting year 0 hr	
Owner's Contact Information Fax Number (999-999-9999) 204-444-3283 E-Mail Address: KELLYCOPP@MYMTS.NET		Other aerial work 0 hr	
Aircraft Make PIPER	Aircraft Model PA22 150 X	Aircraft Serial Number 223388	
Empty Weight Last Actual 1164 lb	Which landing gear configuration? <input checked="" type="checkbox"/> Wheels <input type="checkbox"/> Floats <input type="checkbox"/> Skis Date (yyyy-mm-dd) 2008-06-12	Last Amended 0 lb Which landing gear configuration? <input checked="" type="checkbox"/> Wheels <input type="checkbox"/> Floats <input type="checkbox"/> Skis Date (yyyy-mm-dd)	
Max. permissible take-off weight 2000 lb	Which landing gear configuration? <input checked="" type="checkbox"/> Wheels <input type="checkbox"/> Floats <input type="checkbox"/> Skis		
Engine Make AVCO LYCOMING	Propeller Make		
Engine Model 1 O-320 X	Engine Serial Number 1 L576527	Propeller Model 1	Propeller Serial Number 1
Engine Model 2	Engine Serial Number 2	Propeller Model 2	Propeller Serial Number 2
Engine Model 3	Engine Serial Number 3	Propeller Model 3	Propeller Serial Number 3
Engine Model 4	Engine Serial Number 4	Propeller Model 4	Propeller Serial Number 4
Ski Make	Ski Model	Float Make	Float Model
Owner Remarks		AIRCRAFT OUT OF SERVICE <input checked="" type="checkbox"/> Note: Hours flown and TSN must be completed up to the date of out of service This aircraft will be/has been out of service from (yyyy-mm-dd) 2020-10-01 Estimated date of return to service (yyyy-mm-dd) 2027-01-01 Reason (optional) OTHER/UNKNOWN	

I CERTIFY THAT THE INFORMATION SUPPLIED IS CORRECT

Transport Canada (AARDA)
330 Sparks Street
Place de Ville, Tower C
Ottawa, Ontario K1A 0N8

Name of owner or authorized agent (please print) _____ Date (yyyy-mm-dd) _____

Signature of owner or authorized agent _____

ANNUAL AIRWORTHINESS INFORMATION REPORT (AAIR)

Under the Aviation Regulations (CARs), Part V, Subpart 1, and Chapter 501 of the Airworthiness Manual, the owner of a Canadian aircraft, other than the owner of a light aircraft, shall submit to the Minister no later than the due date an Annual Airworthiness Information Report (AAIR). For questions related to this form, please contact Transport Canada Headquarters at 1-888-663-3639 (Option 1) or by email at tc.aair-rama.tc@tc.gc.ca

AIRCRAFT REGISTRATION

For all changes to your Aircraft Registration including but not limited to: change of address, aircraft sold, aircraft destroyed, aircraft retired, etc., contact your TC Regional Office at 1-800-305-2059, option 1 for English, 2 for Aircraft Registration and then select the appropriate region. **Indicating this type of information on your AAIR form (hard copy or online) will not update your Aircraft Registration.**

HOW TO SUBMIT THE AAIR

The AAIR can be submitted in one of two ways:

1. Fill in the attached form and return it to Transport Canada (TC), following the instructions of sections A, B and C below; or
2. Submit it electronically via the Internet, following the instructions of section D below.

A. PROVIDE DATA IN THE BLANK SPACES

1. Date of the most recent annual or 100-hour inspection, and the name & number of the AMO or AME who/which conducted and certified the inspection. In the case of an amateur-built or owner maintained aircraft, the name of the owner if the owner performed the inspection.

Note: Not required for aircraft operated pursuant to CAR IV or CAR VII.

Note: The AAIR due date and the aircraft's inspection date are not related.

2. Indicate (yes or no) whether the aircraft was significantly damaged since the last report. If YES, the date the aircraft was damaged and, if applicable, the date of the repair certification.
3. Aircraft total hours flown since new – to December 31st of the reporting year, and aircraft hours flown from January 1st to December 31st of the reporting year.
Note: Air operators and flight training units may provide a consolidated breakdown of the total training hours and the total hours flown for other aerial work activities (specialized work) in the reporting year.
4. Provide the aircraft owner's 24-hour fax number and/or e-mail address to which TC can forward notifications of applicable Airworthiness Directives (ADs) and Civil Aviation Safety Alerts (CASAs), as well as to address matters related to AAIRs.

B. AIRCRAFT OUT OF SERVICE

Subject to certain conditions, an AAIR is not required for an aircraft that is out of service (not flown) for all of a complete calendar year. When applicable, an owner can claim this privilege by indicating on this form:

1. Check the box in the AIRCRAFT OUT OF SERVICE section; and
2. Indicate the date the aircraft will be/has been out of service from; and
3. Indicate the estimated date of return to service. If unknown, indicate an approximate date.

Note: If the aircraft is declared out of service part way into a calendar year, an AAIR will be sent the following year in order to capture the hours flown, and other related data, from January 1st up to the date the aircraft was declared out of service.

C. MAILING INSTRUCTIONS

1. Check that all necessary data has been supplied.
2. Print the form, sign and date the lower right-hand corner.
3. Mail the form to:
Transport Canada (AARDA)
330 Sparks St, Place de Ville, Tower C
Ottawa, Ontario K1A 0N5
4. Add sufficient postage and ensure that the envelope is postmarked no later than the due date.

D. INTERNET SUBMISSION INSTRUCTIONS

1. Using your AAIR User ID, log in to the Continuing Airworthiness Web Information System (CAWIS) at www.tc.gc.ca/cawis-swimn (Use of your AAIR User ID is deemed to be the equivalent of your signature when submitting an AAIR via the Internet.)

Note: User IDs are assigned by TC and will remain the same each year. If you do not have a User ID or have forgotten it, you can retrieve your User ID and create a password by selecting the "Forgot User ID or Password" link on the AAIR Login Page.

2. Check and update all existing data, and enter the required new data following the guidance of sections A and B above.
3. Click the "Submit" button at the bottom of the page to send your completed AAIR to Transport Canada. The page will refresh, confirming your AAIR was successfully submitted.
4. Click the "Print" button at the bottom of the page if you would like to keep a hardcopy for your records.

Thank You.

The AAIR has been successfully submitted:

2021-03-26

Date (yyyy-mm-dd)

If there are any discrepancies with the information you have provided, Transport Canada will be in contact with you. It is recommended that you now print or save this acknowledgement as proof of compliance with CAR 501.

Float Make

Float Model

AIRCRAFT OUT OF SERVICE
☒

Note: Hours flown and TSN must be completed up to the date of out of service.

This aircraft will be/has been out of service from (yyyy-mm-dd)

2020-10-01

Estimated date of return to service (yyyy-mm-dd)

2027-01-01

Reason (optional)

OTHER/UNKNOWN

I CERTIFY THAT THE INFORMATION SUPPLIED IS CORRECT

Name of owner or authorized agent (please print)

Date (yyyy-mm-dd)

B R0E 1J2

TERMS Due on receipt

ACTIVITY	DESCRIPTION	TAX	QTY	RATE	AMOUNT
121.5 ELT	ELT tested and recertified in accordance with cars 571 Appendix G. p/n AK-450, s/n 465146. Battery Due November2020. C-FZNR	GST/PST MB	1	52.50	52.50
scribed above has been performed in accordance andards of airworthiness.					52.50
inspector					2.63
AMO 6 85					3.68
SUBTOTAL					58.81
GST @ 5%					
PST (MB) @ 7%					
TOTAL					\$58.81

Authorized Release Certificate

1. Approving Civil Aviation Authority/Country Transport Canada		2. AUTHORIZED RELEASE CERTIFICATE FORM ONE			3. Form Tracking No. 10373
4. Organization Name and Address Saskatoon Avionics					5. Work Order/Contract/Invoice 10373
28 Airport Crescent, Saskatoon, Saskatchewan S7L 6G9, Canada 306-244-4252					
6. Item 1	7. Description ELT	8. Part No AK-450	9. Qty 1	10. Serial/Batch No. 465146	11. Status/work Tested/Inspected
12. Remarks ELT tested and inspected in accordance with CAR 571, Appendix G. Battery due Nov/2020.					
13a. Certifies that the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> CARS 571.10 Maintenance Release <input type="checkbox"/> Other Regulation Specified in block 12		
<input type="checkbox"/> approved design data and are in condition for safe operation			Certifies that unless otherwise specified in block 12, the work identified in block 11 and described in block 12, has been performed in compliance with the Canadian Aviation Regulations.		
<input type="checkbox"/> non approved design data specified in block 12					
13b. Signature		13c. Approved Organization Number		14b. Signature	
				AMO 6-85	
13d. Name		13e. Date (dd/mm/yyyy)		14d. Name	
				Vern McMahon M431274	
				14e. Date (dd/mm/yyyy) 08/10/2019	

Installer Responsibilities

This certificate does not constitute authority to install.
Installers working in accordance with the national regulations of a country other than that specified in block 1 must ensure that their regulations recognize certifications from the country specified.
Statements in blocks 13a or 14a do not constitute



Transport Transports
Canada Canada

CERTIFICATE OF REGISTRATION OF AIRCRAFT
CERTIFICAT D'IMMATRICULATION DE L'AERONEF

Nationality and Registration Marks Marques de nationalité et d'immatriculation C - FZNR	Aircraft or Kit Manufacturer and Model Designation - Constructeur et désignation de l'aéronef ou du kit de l'aéronef Piper Aircraft Corporation	
Purpose - Objet Private - Privé	PA-22-150X	Aircraft Serial Number - Numéro de série de l'aéronef 22-3388X
Registered Owner(s) - Propriétaire(s) enregistré(s) Kelly L Copp		
Address - Adresse 92 Elm Drive Oakbank MB R0E 1J2 CANADA		
<small>Whereas it has been duly declared that the above-mentioned aircraft is not registered in any other State, it is hereby certified that it has been duly entered on the Register of Canada in accordance with the Convention on International Civil Aviation dated the 7th of December, 1944, and the <u>Canadian Aviation Regulations</u>. Attendu que l'aéronef mentionné ci-dessus n'a pas été immatriculé dans un autre État, je certifie par la présente qu'il a été inscrit au registre du Canada, conformément à la Convention relative à l'Aviation civile internationale signée le 7 décembre 1944 et du <u>Règlement de l'aviation canadien</u>.</small>		
1 September/septembre 2011 Owner Registration Date Date d'immatriculation du propriétaire	1 September/septembre 2011 Certificate Issue Date Date de délivrance du Certificat	 For the Minister of Transport - Pour le ministre des Transports

26-0478 (0106-02) FF 1.1

This Certificate is not a Certificate of Title - Ce certificat n'est pas un certificat de titre

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Canada

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PART 1 - PARTIE 1



Transport Canada
Transports Canada

APPLICATION FOR REG'N OF AIRCRAFT (NOT FOR ULTRA-LIGHT OR ADVANCED ULTRA-LIGHT AEROPLANES) DEMANDE D'IMM. D'AÉRONEF (PAS POUR LES AVIONS ULTRA-LÉGERS OU ULTRA-LÉGERS DE TYPE ÉVOLUÉ)

Application is hereby made for registration, in accordance with the Canadian Aviation Regulations, the particulars of which are contained herein (see Part 4 on the back of this form).
certificat de registration pour directions) : En conformité avec les dispositions du Règlement de l'aviation canadien, la présente est soumise pour l'immatriculation, dont les détails sont décrits (voir la partie 4 au verso du certificat d'immatriculation intérimaire pour les instructions).

A. GENERAL INFORMATION - INFORMATION GÉNÉRALE

Model Designation - Désignation du modèle

PA-22-150X

Serial Number - Numéro de série

22-3388X

Proposed Purpose - But proposé

☐ Private - Privé

☐ State - État ☐ Commercial

Proposed Base of Operation
Base d'utilisation proposée

Registration Marks - Marques d'immatriculation

C - FZNR

Maximum Take-off Weight
Masse maximale admissible au décollage

☐ kg ☐ lbs - lvs

Category of Engine - Catégorie de moteur

☐ Piston
À pistons

☐ Turbo Jet
Turboréacteur

☐ Turbomoteur
Turboréacteur

☐ Turbo Prop
Turbopropulseur

☐ Turbo Fan
Turbosoufflante

☐ Other
Autre

Number of Engines
Nombre de moteurs

B. PARTICULARS OF APPLICANT(S) - PRÉCISIONS CONCERNANT LE(S) DEMANDEUR(S)

For more than one owner applying for registration, attach an addendum with the remaining applicants' names, addresses, citizenship, telephone numbers, pilot licence number and signatures. Pour plus d'un propriétaire demande un certificat d'immatriculation, il faut joindre un addendum lesquelles figureront le nom, l'adresse, la nationalité, le numéro de téléphone, le n° de licence de pilote et la signature de ces demandeurs.

1. Name (if an entity, documentary proof of incorporation is req'd) - Nom (Si un organisme, annexer une pièce justificative de l'incorporation)

Address - Adresse

Daytime Telephone - Téléphone du jour

Pilot Licence Number - No de licence de pilote

Evening Telephone - Téléphone de soirée

Email - Courriel électronique

Facsimile - Télécopieur

Citizenship - Nationalité

C. TO BE COMPLETED IF APPLICANT IS A CANADIAN ENTITY - À REMPLIR LORSQUE LE DEMANDEUR EST UN ORGANISME CANADIEN

Formation Date - Date de l'incorporation

D/U M Y/A

Corporation/Entity Number (where appl.)
Numéro de l'organisme (le cas échéant)

Incorporated - Constitué

☐ Federally
Fédéral ☐ Provincially
Provincial

The above entity meets the definition of a 'Canadian entity' (a Canadian entity is an entity that is incorporated or formed under the laws of Canada or a province, that is controlled in fact by Canadians and of which at least 75% of the voting interests are owned and controlled by Canadians).

☐ Yes
Oui

L'organisme mentionné ci-dessus satisfait à la définition d'un «organisme canadien» (un organisme canadien est un organisme constitué au Canada sous le régime de lois fédérales ou provinciales et contrôlé de fait par des Canadiens, dont au moins 75% des actions assorties du droit de vote sont détenues et contrôlées par des Canadiens).

☐ No
Non

Name and title of person authorized to sign on behalf of the Entity - Nom et titre de la personne autorisée à signer au nom de l'organisme

D. I CERTIFY THAT - JE CERTIFIE (QUE)

1. (A) ☐ I am/we are the sole person/entity that has legal custody and control of the aircraft described herein.
or - du Être le(s) seul(s) propriétaire(s) avec la garde et la responsabilité légales de l'aéronef décrit dans la présente formule.
- (B) ☐ I am authorized by the owner(s) of the aircraft described herein to make this application (documentary proof is required).
Être dûment autorisé par le(s) propriétaire(s) à soumettre la présente demande d'immatriculation (annexer la pièce justificative en question).
2. I am/we are qualified to be the owner(s) of a Canadian registered aircraft in accordance with section 202.15 of the Canadian Aviation Regulations.
Je suis (nous sommes) qualifié(s) pour être le(s) propriétaire(s) d'un aéronef immatriculé au Canada conformément à l'article 202.15 du Règlement de l'aviation canadien.

It is an offence under section 7.3(1)(a) of the Aeronautics Act to knowingly make a false declaration.
Faire sciemment une fausse déclaration est une infraction en vertu de l'alinéa 7.3(1)(a) de la loi sur l'aéronautique.

1. Signature of Applicant - Signature du demandeur

Date

E. RECOMMENDATION (FOR TRANSPORT CANADA USE ONLY) - RECOMMANDATION (POUR TRANSPORTS CANADA SEULEMENT)

Recommendation for Registration (Name and Position) - Recommandation pour l'immatriculation (Nom et titre)

Date of Recommendation - Date de recommandation

PART 2 - PARTIE 2



Transport Canada
Transports Canada

CERTIFICATE OF REGISTRATION OF AIRCRAFT CERTIFICAT D'IMMATRICULATION DE L'AERONEF

Nationality and Registration Marks Marques de nationalité et d'immatriculation C - FZNR	Aircraft or Kit Manufacturer and Model Designation - Constructeur et désignation de l'aéronef ou du kit de l'aéronef Piper Aircraft Corporation PA-22-150X
Purpose - Objet N/A - Non applicable	Aircraft Serial Number - Numéro de série de l'aéronef 22-3388X
Registered Owner(s) - Propriétaire(s) enregistré(s)	

Interim Certificate of Registration Certificat d'immatriculation intérimaire

Address - Adresse

Whereas it has been duly declared that the above-mentioned aircraft is not registered in any other State, it is hereby certified that it has been duly entered on the Register of Canada in accordance with the Convention on International Civil Aviation dated the 7th of December, 1944, and the Canadian Aviation Regulations.

Attendu que l'aéronef mentionné ci-dessus n'a pas été immatriculé dans un autre Etat, je certifie par la présente qu'il a été inscrit au registre du Canada, conformément à la Convention relative à l'Aviation civile internationale signée le 7 décembre 1944 et du Règlement de l'aviation canadien.

1 September/septembre 2011

Owner Registration Date
Date d'immatriculation du propriétaire

Change of Ownership/Information Date
Date de changement de propriétaire/information

For the Minister of Transport - Pour le ministre des Transports

Harold Ricard

26-0478 (0106-02) FF 1.1

This Certificate is not a Certificate of Title - Ce certificat n'est pas un certificat de titre

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Canada

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KEEP ON BOARD AIRCRAFT - LAISSEZ À BORD DE L'AÉRONEF
DO NOT RETURN TO TRANSPORT CANADA - NE PAS RETOURNER À TRANSPORTS CANADA

SEE REVERSE SIDE FOR ACTIVATING INTERIM REGISTRATION - VOIR AU VERSO POUR IMMATRICULATION INTÉRIMAIRE

REGIONAL ADDRESSES

Regional Manager, General Aviation, Transport Canada Civil Aviation
Pacific Region

620-800 Burrard St., Vancouver BC V6Z 2J8

Telephone: (604) 666-5575 Facsimile: (604) 666-4839

Regional Manager, General Aviation, Transport Canada Civil Aviation
Prairie and Northern Region

1100-9700 Jasper Avenue, Edmonton AB T5J 4E6

Telephone: (780) 495-5249 Facsimile: (780) 495-7449

Regional Manager, General Aviation, Transport Canada Civil Aviation
Prairie and Northern Region

P.O. Box 8550, 344 Edmonton Street, 2nd Floor, Winnipeg MB R3C 0P6

Telephone: (204) 983-4341 Facsimile: (204) 984-2069

Regional Manager, General Aviation, Transport Canada Civil Aviation
Ontario Region

Yonge St., Suite 300, Toronto ON M2N 6A5

Telephone: (416) 952-0215 Facsimile: (416) 952-0196

Regional Manager, General Aviation, Transport Canada Civil Aviation
Quebec Region

1000, rue Saint-Jacques, Suite 200, Montréal QC H4Y 1G7

Telephone: (514) 633-3580 Facsimile: (514) 633-3585

Regional Manager, General Aviation, Transport Canada Civil Aviation
Atlantic Region

42, Heritage Court, 95 Foundry Street, Moncton NB E1C 8K6

Telephone: (506) 851-7131 Facsimile: (506) 851-2563

ADRESSES RÉGIONALES

Gestionnaire régional, Aviation générale, Transports Canada, Aviation civile
Région du Pacifique

620-800, rue Burrard, Vancouver BC V6Z 2J8

Téléphone: (604) 666-5575 Télécopieur: (604) 666-4839

Gestionnaire régional, Aviation générale, Transports Canada, Aviation civile
Région des Prairies et Nord

1100-9700 Av. Jasper, Edmonton AB T5J 4E6

Téléphone: (780) 495-5249 Télécopieur: (780) 495-7449

Gestionnaire régional, Aviation générale, Transports Canada, Aviation civile
Région des Prairies et Nord

C.P. 8550, 344, rue Edmonton, 2^e étage, Winnipeg MB R3C 0P6

Téléphone: (204) 983-4341 Télécopieur: (204) 984-2069

Gestionnaire régional, Aviation générale, Transports Canada, Aviation civile
Région de l'Ontario

4900, rue Yonge, pièce 300, Toronto ON M2N 6A5

Téléphone: (416) 952-0215 Télécopieur: (416) 952-0196

Gestionnaire régional, Aviation générale, Transports Canada, Aviation civile
Région du Québec

Bureau régionale

700, Leigh Capreol, pièce 2033, Dorval QC H4Y 1G7

Téléphone: (514) 633-3580 Télécopieur: (514) 633-3585

Gestionnaire régional, Aviation générale, Transports Canada, Aviation civile
Région de l'Atlantique

C.P. 42, Place Heritage, 95, rue Foundry, Moncton NB E1C 8K6

Téléphone: (506) 851-7131 Télécopieur: (506) 851-2563

PART 3 - PARTIE 3

Transport Canada
Transports Canada**NOTIFICATION OF CHANGE OF OWNER(S)**
AVIS DE CHANGEMENT DE(DES) PROPRIÉTAIRE(S), being the registered owner of aircraft
, propriétaire enregistré de l'aéronef

Aircraft Manufacturer - Constructeur de l'aéronef

Manufacturer's Designation of Aircraft - Désignation du constructeur de l'aéronef

Aircraft Serial Number - Numéro de série de l'aéronef

Marks - Marques

this date

this date

Transfer ownership of the above aircraft to

Fais un transfert de propriété à

Name of New Owner - Nom du nouveau propriétaire

Address - Adresse

City - Ville

Province

Postal Code - Code postal

This document is to assist the registered owner in meeting the regulatory requirements to advise the Minister of a change of aircraft owner. This card does not establish legal custody and control of the aircraft.

Ce document vise à aider le propriétaire enregistré à satisfaire aux exigences réglementaires en vue d'aviser le ministre d'un changement de propriétaire. Cette carte n'établit pas la garde et la responsabilité légales de l'aéronef.

Signature of Registered Owner
Signature du propriétaire enregistré

Canada

5-0519 (0603-02)

PA22 158 X

With landing gear configuration
WHEELSWith landing gear configuration
WHEELSWith landing gear configuration
WHEELS

Aircraft Serial Number

ATTACHMENT C
EXEMPTION FROM THE *CANADIAN AVIATION REGULATIONS*
(AS LISTED IN THE APPENDIX TO THIS EXEMPTION)

"FLIGHT PERMIT - SPECIFIC PURPOSE - OWNER MAINTENANCE"

Pursuant to subsection 5.9(2) of the *Aeronautics Act*, and after taking into account that the exemption is in the public interest and is not likely to affect aviation safety, I hereby exempt the owner(s) of aircraft in respect of which a Flight Permit - Specific Purpose - Owner Maintenance has been issued, from the application of the following provisions, subject to the conditions contained in this exemption:

Sections 507.10, 571.04, 571.07, 571.09, 571.12, 571.13; Subsections 571.02 (3), 571.06 (1), 571.08 (1), 571.11 (1), 571.11 (4), 605.84 (1) and Paragraphs 602.59(1)(a) and 605.92 (1)(b) of the *Canadian Aviation Regulations*, details of which are found in the Appendix of this Exemption.

PURPOSE

The purpose of this exemption is to provide interim relief to owners of small aeroplanes pending the implementation in the *Canadian Aviation Regulations* of a new flight authority, the Special Certificate of Airworthiness - Owner Maintenance, which will be issued for recreational purposes only.

The exemption frees the owner(s) from the need to comply with the requirements of the regulations cited in this exemption. Not only does it permit the owner(s) of a small aeroplane to perform the maintenance on his/her/their aircraft, effect modifications and repairs without Transport Canada approval, and install uncertified parts, the exemption also authorizes the owner(s) to sign the maintenance release for all maintenance performed on his/her aircraft.

APPLICATION

This exemption applies to the owner(s) of a small aeroplane in respect of which a Flight Permit - Specific Purpose - Owner Maintenance has been issued.

CONDITIONS

This exemption is subject to the following conditions:

1. A Flight Permit - Specific Purpose - Owner Maintenance has been issued in respect of the aircraft and is in force;
2. The letter "X" is permanently etched, engraved or stamped at the end of the model designation and serial number on the identification plate of the subject aircraft as well as on the identification information of each engine, propeller and life-limited part installed on this aircraft;
3. A new Certificate of Registration, reflecting the revised model designation and serial number of the aircraft, has been issued in respect of the aircraft;
4. The aeroplane shall have, on the side of the fuselage, in a position that is readily visible to persons entering the aircraft, in letters at least 10 mm (3/8 in.) high and of a colour contrasting with the background, in both official languages, a placard containing the following statement:

ATTACHMENT C
EXEMPTION FROM THE *CANADIAN AVIATION REGULATIONS*
(AS LISTED IN THE APPENDIX TO THIS EXEMPTION)

"FLIGHT PERMIT - SPECIFIC PURPOSE - OWNER MAINTENANCE"

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3. A new Certificate of Registration, reflecting the revised model designation and serial number of the aircraft, has been issued in respect of the aircraft;
4. The aeroplane shall have, on the side of the fuselage, in a position that is readily visible to persons entering the aircraft, in letters at least 10 mm (3/8 in.) high and of a colour contrasting with the background, in both official languages, a placard containing the following statement:

ATTACHMENT C

WARNING

THIS AIRCRAFT IS OPERATED
WITHOUT A CERTIFICATE OF AIRWORTHINESS
AND DOES NOT COMPLY WITH INTERNATIONALLY
RECOGNIZED AIRWORTHINESS STANDARDS.
YOU FLY IN THIS AIRCRAFT AT YOUR OWN RISK

AVIS

CET AÉRONEF EST EXPLOITÉ
SANS CERTIFICAT DE NAVIGABILITÉ
ET N'EST PAS CONFORME AUX NORMES
DE NAVIGABILITÉ INTERNATIONALES RECONNUES.
VOUS PRENEZ PLACE À BORD À VOS PROPRES RISQUES

5. The person who performs a repair or modification on an aeronautical product or signs a maintenance release in respect of such a repair or modification, shall ensure that, as a minimum, "acceptable technical data", as defined in Section 571.06 of the *Airworthiness Manual*, is used to perform the repair or modification;
6. The person who signs a maintenance release in accordance with Section 571 of the *Airworthiness Manual* is the holder of pilot license or permit and an owner of the aircraft; and
7. A copy of this exemption shall be carried on board the aircraft during operations.

VALIDITY

This exemption is in effect until the earliest of the following:

- (a) EST at midnight December 31, 2004;
- (b) The date on which an amendment to the appropriate provisions of the *Canadian Aviation Regulations* comes into effect, at which time the owner(s) shall submit an application for a new flight authority: Special Certificate of Airworthiness - Owner Maintenance;
- (c) The date on which any one of the conditions set out in this exemption is breached; or
- (d) The date on which this exemption is cancelled in writing by the Minister where he or she is of the opinion it is no longer in the public interest, or is likely to affect aviation safety.

Dated at Winnipeg, Manitoba, this 21 day of July 2000, on behalf
of the Minister of Transport.

Art LaFlamme
Director General
Civil Aviation



APPENDIX TO ATTACHMENT C

CAR 507.10 – Persons Who May Attest to Condition and Conformity

No person shall make a declaration of an aircraft's condition or conformity to its certified type design for the purpose of obtaining a flight authority other than the holder of an aircraft maintenance engineer (AME) licence issued pursuant to Part IV or

- (a) in the case of a new aircraft, an authorized representative of the manufacturer; or
- (b) in the case of an amateur-built aircraft, the owner of the aircraft.

CAR 571.02 (3) – Maintenance Performance Rules

Except where the work is performed in respect of an amateur-built aircraft, no person shall supervise, or perform without supervision, an inspection using a method set out in column I of an item of Schedule I to this Subpart, unless the person holds the personnel certification set out in column II of that item.

CAR 571.04 – Specialized Maintenance

No person shall perform the specialized maintenance set out in Schedule II to this Subpart on an aeronautical product other than amateur-built aircraft, except in accordance with

- (a) a maintenance policy manual (MPM) established by the holder of an approved maintenance organization (AMO) certificate issued pursuant to Section 573.02 with a rating of a category appropriate to the work to be performed; or
- (b) a foreign document equivalent to an MPM established by a maintenance organization approved under the laws of a state that is party to an agreement with Canada, and the agreement provides for recognition of the work to be performed.

CAR 571.06 (1) – Repairs and Modifications

A person who performs a major repair or major modification, or signs a maintenance release in respect of such a repair or modification, shall ensure that the major repair or major modification conforms to the requirements of technical data that have been approved or the use of which has been approved within the meaning assigned to the term "Approved Data" in Section 571.06 of the *Airworthiness Manual*.

CAR 571.07 – Installation of a New Part

- (1) No person shall install a new part on an aeronautical product unless the part meets the standards of airworthiness applicable to the installation of new parts and, subject to subsections (2) and (3), has been certified in accordance with Chapter 561 of the *Airworthiness Manual*.
- (2) No certification referred to in subsection (1) is required where
 - (a) a new part is a foreign-manufactured part that is certified pursuant to an agreement with Canada, which agreement provides for the acceptance of export airworthiness certification;
 - (b) a new part is a foreign-manufactured part that is obtained from a manufacturer holding a type design recognized in Canada and the part is certified in accordance with the laws of the state of manufacture;
 - (c) a new part, whose accompanying documentation has been verified, has been inspected in accordance with the requirements of Chapter 571 of the *Airworthiness Manual*;
 - (d) a new part is installed on an amateur-built aircraft; or
 - (e) a part is made in accordance with subsection 571.06(4).
- (3) No certification referred to in subsection (1) is required in respect of a new part that bears markings identifying it as a part specified in the type design and that

APPENDIX TO ATTACHMENT C

- (a) conforms to a standard that is recognized as a national or international standard for use on the aeronautical product;
- (b) was originally designed and manufactured for non-aeronautical use, on condition that it has been approved for use on the aeronautical product, in accordance with the type certificate; or
- (c) was manufactured under a Parts Manufacturer Approval issued by the Federal Aviation Administration (United States), where that part meets the installation criteria specified in Chapter 571 of the *Airworthiness Manual*.

CAR 571.08 (1) – Installation of Used Parts

No person shall install a used part on an aeronautical product, other than an amateur-built aircraft, unless the part meets the standards of airworthiness applicable to the installation of used parts stated in Chapter 571 of the *Airworthiness Manual* and

- (a) is an airworthy part that has been removed from an aircraft for immediate installation on another aircraft;
- (b) is an airworthy part that has undergone maintenance for which a maintenance release has been signed pursuant to paragraph 571.11(2)(c); or
- (c) has been inspected and tested to ensure that the part conforms to its design standards and is in a safe condition, and a maintenance release has been signed to that effect.

CAR 571.09 - Installation of Life-limited Parts

- (1) No person shall install a used life-limited part on an aeronautical product unless the part meets the standards of airworthiness applicable to the installation of life-limited parts and
 - (a) the technical history of the part within the meaning of Section 571.09 of the *Airworthiness Manual* is available to show that the time in service authorized for that part in the type certificate governing the installation has not been exceeded; and
 - (c) the history referred to in paragraph (a) is incorporated into the technical record for the aeronautical product on which the part is installed.
- (2) No person shall install a used part that has been subject to a life limitation in a place other than that from which it was removed unless the part is installed
 - (a) in the same or in an identical position on another aeronautical product bearing the same part number as that from which the part was removed; or
 - (b) in conformity with the requirements in respect of technical data that have been approved or the use of which has been approved within the meaning of Section 571.09 of the *Airworthiness Manual*.

CAR 571.11 (1) – Persons Who May Sign a Maintenance Release

Except as provided in subsections (2) and (4), no person other than the holder of an aircraft maintenance engineer (AME) licence issued pursuant to Part IV, specifying a rating appropriate to the aeronautical product being maintained, shall sign a maintenance release in accordance with Chapter 571 of the *Airworthiness Manual*.

CAR 571.11 (4) – Persons Who May Sign a Maintenance Release

A person who is not the holder of an AME licence specifying a rating appropriate to the aeronautical product being maintained may sign a maintenance release where the person holds a restricted certification authority issued in accordance with Chapter 566 of the *Airworthiness Manual*, in respect of a particular case specified on the restricted certification authority.

APPENDIX TO ATTACHMENT C

CAR 571.12 – Reporting Major Repairs and Major Modifications

A person who performs a major repair or major modification on an aeronautical product or installs on an aircraft a part that has undergone a major repair or major modification shall report the action to the Minister in accordance with the procedures specified in section 571.12 of the *Airworthiness Manual*.

CAR 571.13 – Installation of Parts (General)

Subject to sections 571.07 to 571.09, no person shall install a part on an aeronautical product unless the part is

- (a) inspected and its accompanying documentation verified in accordance with a procedure that ensures that the part conforms to its design standards with regard to safety of the aircraft; and
- (b) installed in accordance with the requirements of section 571.13 of the *Airworthiness Manual*.

CAR 602.59 (1)(a) – Equipment Standards

Subject to subsection (2), no person shall operate an aircraft unless the operational and emergency equipment carried on board the aircraft...

meets the applicable standards specified in the *Airworthiness Manual*.

CAR 605.84 (1) – Aircraft Maintenance - General

Subject to subsections (3) and (4), no person shall conduct a take-off, or permit a take-off to be conducted in an aircraft that is in the legal custody and control of the person, unless the aircraft is maintained in accordance with

- (a) any airworthiness limitations applicable to the aircraft type design;
- (b) the requirements of any airworthiness directives issued by the Minister pursuant to section 593.02;
- (c) except as provided in subsection (2), the requirements of any notices equivalent to airworthiness directives issued by
 - (i) the competent authority of the foreign state that, at the time the notice was issued, is responsible for the type certification of the aircraft, engine, propeller or appliance, or
 - (ii) for an aeronautical product in respect of which no type certificate has been issued, the competent authority of the foreign state that manufactured the aeronautical product.

CAR 605.92 (1)(b) – Requirements to Keep Technical Records

Every owner of an aircraft shall keep the following technical records in respect of the aircraft:

subject to (2) and (3), a separate technical record for the airframe, each installed engine and each variable pitch propeller.



Transport Canada / Transports Canada

OWNER MAINTENANCE

SPECIAL CERTIFICATE OF AIRWORTHINESS

CERTIFICAT DE NAVIGABILITÉ SPÉCIAL

1. Nationality and Registration Marks Marques de nationalité et d'immatriculation C-FZNR		2. Aircraft Manufacturer and Model Constructeur et modèle de l'aéronef Piper Aircraft Corporation Piper PA-22-150X		3. Aircraft Serial Number Numéro de série de l'aéronef 22-3388X	
4. Engine Manufacturer – Constructeur du moteur Avco Lycoming		5. Engine Model – Modèle de moteur O-320		6. Maximum Permissible Take-off Weight Masse maximale admissible au décollage Kg 2000 lbs.	
7. Classification <input checked="" type="checkbox"/> provisional provisoire <input type="checkbox"/> restricted restreinte <input type="checkbox"/> amateur-built construction amateur <input type="checkbox"/> limited limitée					
8. This Certificate of Airworthiness is issued pursuant to the Aeronautics Act and certifies that, as of the date of issuance, the aircraft to which it was issued has been inspected, found to be in a condition for safe operation. 8. Ce certificat de navigabilité est délivré en vertu de la Loi sur l'aéronautique et certifie qu'à la date de délivrance dudit certificat, l'aéronef visé par ce certificat, a été inspecté, est conforme à son homologation de type et est apte à voler en toute sécurité.					
9. This aircraft has not been shown to comply with the International Civil Aviation Organization airworthiness standards. Therefore, approval of the foreign airworthiness authority is required prior to flight over their territory. 9. Il n'a pas été démontré que cet aéronef est conforme aux normes de navigabilité de l'Organisation de l'Aviation civile internationale. Il faut par conséquent, obtenir la permission de l'Autorité de la navigabilité du pays étranger concerné avant de survoler leur territoire.					
10. Operating limitations dated <u>July 18, 2002</u> are a part of this certificate. 10. Les limites d'utilisation en date du <u>juillet 18/02</u> font partie du présent certificat.					
11. Unless suspended or cancelled in accordance with the Aeronautics Act , this certificate shall remain in force as long as the aircraft identified above is maintained and certified in accordance with the Airworthiness Manual. 11. Conformément à la Loi sur l'aéronautique , à moins qu'il n'y ait suspension ou annulation, le présent certificat reste en vigueur tant que l'aéronef ci-haut mentionné est entretenu et certifié conformément au Manuel de navigabilité.					
<u>July 18, 2002</u> Date of Issue – Date de délivrance		<u>Noel Blais</u> For the Minister of Transport – Pour le ministre des Transports		Seal Sceau	

24-0074 (12-90)

Canada

United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number SA849GL

This certificate, issued to EAA Aviation Foundation, Inc.
Wittman Airfield
Oshkosh, Wisconsin 54903-3065

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 3 of the Civil Air

Regulations. (See Aircraft Specification No. 1A6 for complete certification basis.)

Original Product — Type Certificate Number: 1A6

Make: Piper Aircraft Corporation

Model: PA-22, PA-22-108, -135, -150, -160,
PA-22S, -135, -150, and -160

Description of Type Design Change:

Modify airplane to fly on unleaded automotive gasoline, 87 minimum antiknock index, per ASTM Specification D-439. The engine installed must conform to a Supplemental Type Certificate approving the use of unleaded automotive gasoline as above for the engine model installed in the aircraft.

Limitations and Conditions: 1. Airplane Flight Manual Supplement dated November 9, 1984, or subsequent FAA approved revision is required. 2. FAR 43 combined with the Airplane Flight Manual Supplement is adequate to ensure continued airworthiness of this modification. 3. This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationships between this change and any of those other previously approved modifications will introduce no adverse effect on the airworthiness of that aircraft.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: September 27, 1984

Date received:

Date of issuance: November 9, 1984

Date amended: June 28, 1985



By direction of the Administrator
Donald P. Michael

W. F. HORN

(Signature)

Manager, Chicago Aircraft Certification
Office, ACE-115C

Central Region (Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number SE800GL

This certificate, issued to EAA Aviation Foundation, Inc.
Wittman Airfield
Oshkosh, Wisconsin 54903-3065

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 13 of the Civil Air Regulations. (See Type Certificate Data Sheet E-274 for complete certification basis.)

Original Product — Type Certificate Number: E-274
Make: Textron Lycoming
Model: O-320A, C, E (See attached Continuation Sheet for additional models.)

Description of Type Design Change:

Add the following approved fuel: unleaded automotive gasoline, 87 minimum antiknock index, per ASTM Specification D-439 of any Volatility Class, A through E.

Limitations and Conditions: This approval should not be extended to other engines of this model on which other previously approved modifications are incorporated, unless it is determined by the installer that the interrelationship between this change and any of those other previously approved modifications will introduce no adverse effect upon the airworthiness of this engine. Specific approval must be obtained for each model aircraft to ensure compatibility with its fuel system. This approval is limited to normally aspirated piston engines in power outputs up to 150 brake horsepower only.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration. (Cont. on page 2 of 2)

Date of application: November 9, 1983

Date received:

Date of issuance: July 30, 1984

Date amended: March 26, 1985; September 16, 1988



By direction of the Administrator

W. F. Horn (Signature)
Manager, Chicago Aircraft
Certification Office, ACE-115C

Central Region (Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.
This certificate may be transferred in accordance with FAR 21.47.

United States of America
Department of Transportation—Federal Aviation Administration
Supplemental Type Certificate
(Continuation Sheet)

Number SE800GL

Dated Amended: September 16, 1988

Model: (Cont'd):

<u>Model</u>	<u>Approval</u>	<u>Type Certificate No.</u>
O-235-C, -E, -H	March 26, 1985	E-223
O-290, O-290-A, -AP, -B -C, -CP, -D, -D2, -D2A, -D2B, -D2C	March 26, 1985	E-229

...END...

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

FAA FORM 8110-2-1 (10-69)

This certificate may be transferred in accordance with FAR 21.47.



DEDICATED TO THE EDUCATION HISTORY AND DEVELOPMENT OF AVIATION

FAA APPROVED

AIRPLANE FLIGHT MANUAL SUPPLEMENT

FOR

PIPER PA-22, PA-22-108,-135,-150,-160,
PA22S-135,-150,-160

REG. NO. CF2NR

SER. NO. 223388

This Supplement must be attached to the FAA Approved Airplane Flight Manual when the airplane is operated using unleaded automotive gasoline in accordance with STC No. SA849GL. The information contained herein supplements or supersedes the basic manual only in those areas listed. For limitations, procedures and performance information not contained in this Supplement, consult the basic Airplane Flight Manual.

FAA APPROVED:

Donald P. Michael
for W. F. Horn, Manager
Chicago Aircraft
Certification Office
Central Region

DATE: November 9, 1984

REVISED: June 28, 1985

Not Valid
Without
Raised Seal

Page 1 of 4

LOG OF REVISIONS

Description of Revision

Log of Revision

Revised Paragraph

Piper Model

108, -135, -1

First Certificate

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LOG OF REVISIONS

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Description of Revision

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Revised Paragraph

Piper Model

108, -135, -1

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Revised Paragraph

Piper Model

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Description of Revision

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Revised Paragraph

Piper Model

108, -135, -1

First Certification

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ated by a b

EAA Aviation Foundation
Wittman Airfield
Oshkosh, Wisconsin 54903-3065

AFM Supplement for
Piper PA-22, PA-22-108
-135, -150, -160, PA-22S-
135, -150, -160

SECTION I - LIMITATIONS

- A. The engine installed in this aircraft must have a valid STC for the use of unleaded automotive gasoline per ASTM Specification D-439, the Antiknock Index must be 87 or more.

B. Placard Required

Located on each wing forward of fuel cap, adjacent to aviation gasoline/fuel tank capacity placard:

Approved Fuel: Unleaded
automotive gasoline, 87
min. antiknock index per
ASTM Spec. D-439.

SECTION II - PROCEDURES

A. Normal

1. Preflight Inspection

It is the pilot-in-command's responsibility to insure that the tank sumps are drained and that the fuel strainer is drained before each flight. It is also his responsibility to take corrective action if water or any other contaminant is found when draining at those points.

2. Fueling with Unleaded Automotive Gasoline

Use the same care as when fueling with aviation fuel to insure that only contaminant-free, water-free fuel enters the tank. It is the responsibility of the pilot-in-command to insure that the fuel conforms to unleaded automotive gasoline per ASTM Specification D-439, 87 minimum antiknock index.

3. Mixing Fuels

Aviation gasoline may be mixed with unleaded automotive gasoline. Any mixture containing unleaded automotive gasoline must be operated in accordance with the placards or precautions established for unleaded automotive gasoline.

1. DO NOT USE GASOLINE CONTAINING ALCOHOL
2. USE LEADED GASOLINE FOR A MINIMUM OF 3 HOURS AFTER AN OVERHAUL AFFECTING THE VALVES—USE UNLEADED AUTOGAS THEREAFTER.

FAA APPROVED

DATE: November 9, 1984
REVISED: June 28, 1985

4. Carburetor Ice

When using unleaded automotive gasoline, the onset of carburetor ice may occur earlier under the same atmospheric conditions than when using 80/87 minimum grade aviation gasoline. There is no change in the techniques for recognizing and correcting for carburetor ice.

5. Engine Operation

Engine operation must conform to Avco Lycoming recommendations except that fuel per this STC is approved.

6. Contamination Control

The following guidance is taken from Advisory Circular (AC) 20-43C, "Aircraft Fuel Control":

Keep fuel tanks full; water condenses on the walls of partially filled tanks and enters the fuel system. Filter all fuel entering the tank. Drain fuel sumps regularly. Periodically inspect and clean all fuel strainers (screens) and occasionally flush the carburetor bowl as recommended by the aircraft manufacturer. The best insurance against fuel problems is to practice good house-keeping in your routine maintenance and be constantly alert.

The operator is referred to this AC for more detail.

B. Emergency

No change.

SECTION III - PERFORMANCE

No change.

THIS DOCUMENT MUST BE KEPT IN THE AIRPLANE AT ALL TIMES

CAA APPROVED
APPROVAL BASIS CAR 3 AND 410
AUGUST 27, 1955
NORMAL CATEGORY
2000 POUNDS GROSS WEIGHT

C.A.A. Identification No. C-EZAR

AIRPLANE FLIGHT MANUAL

JOE SKAVINSKI, DATE
WGM051818

1. Limitations

The following limitations must be observed in the operation of this airplane:

Engine	Lycoming O-320-A2B 150 HP
Engine Limits	For all operations 2700 RPM
Fuel	80/87 Octane Minimum Aviation Gasoline
Propellers	(a) Sensenich M74DM, Fixed Pitch Metal
	74.0" Maximum Diameter
	72.0" Minimum Diameter
	Static Limits: Maximum 2450 RPM
	Minimum 2250 RPM
Power Instruments	Oil Temperature-Unsafe if indicator exceeds
	Red line (245 degrees F.)
	Yellow Arc: Caution (40 degrees F. to
	120 degrees F.)
	Green Arc: Normal Operating Range
	(120 degrees F. to 245 degrees F.)
	Oil Pressure-Unsafe if indicator exceeds
	Red line (100 lbs.) or is below
	the Red line (25 lbs. minimum)
	Yellow Arc: Caution (85 lbs. to 100 lbs.)
	and (25 lbs. to 65 lbs.)
	Green Arc: Normal Operating Range
	(65 lbs. to 85 lbs.)
	Tachometer-Red line: Rated Engine Speed
	Green Arc: 500 RPM to 2700 RPM
	Normal Operating Range
Flap Position	Take-Off 0 Degrees Landing 40 Degrees
Airspeed Limits	
(True Ind. Airspeed)	Normal Category
Maneuvering	112 MPH 97.0 Knots
Max. Cruising Speed	135 MPH 117.0 Knots
Never Exceed	170 MPH 148.0 Knots
Flaps Extended	95 MPH 82.0 Knots

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[Signature]

Flight Load Factors
Max. Positive
Max. Negative
Airplane Loading

3.8

No Inverted Maneuvers Approved
Max. Wgt. (Take-Off and Landing)
2000 Pounds

C. G. Range

(Aft Wing Leading Edge)
{+17.5"} to {+23.0"} at 2000 lbs.
{+12.0"} to {+23.0"} at 1800 lbs.
{+9.5"} to {+23.0"} at 1400 lbs. or less

Maximum Baggage
Allowed

100 Pounds

Note: It is the responsibility of the airplane owner and the Pilot to insure that the airplane is properly loaded. (See Weight and Balance.

Placards:

- (a) On the instrument panel in full view of the Pilot:
 - (1) "Operate in Normal Category in compliance with the Approved Flight Manual. Acrobatics (including spins) prohibited."
- (b) On the Baggage Compartment:
 - (1) "Maximum Baggage 100 Pounds."

Maneuvers

- (a) No acrobatic maneuvers approved for Normal Category Operation.

Airspeed
Instrument
Markings
And Their
Significance

- (a) Radial Red line marks the never exceed speed which is the maximum safe airspeed 170 MPH. (148 Knots)
- (b) Yellow Arc on indicator denotes range of speed in which operations should be conducted with caution and only in smooth air 135 to 170 MPH (117.0 to 148.0 Knots)
- (c) Green Arc denotes normal operating speed range 53 to 135 MPH (39 to 117 Knots)
- (d) White Arc denotes normal operating speed range with flaps extended 49 to 95 MPH (43 to 82 Knots)

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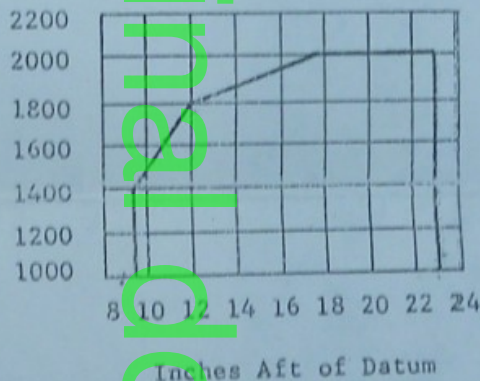
OPERATION LIMITATION

Model PA-22-150, Serial No. 22-2624, 4 PCLM (Normal Category), Approved September 3, 1954.

ENGINE Lycoming O-320-A2B
 FUEL 80/87 Minimum grade aviation gasoline
 ENGINE LIMITS For all operations, 2700 r.p.m. (150 hp.)
 AIRSPEED LIMITS (CAS) V_{ne} (never exceed) 170 m.p.h. (148 knots)
 V_{no} (max. structural cruising) 135 m.p.h. (117 knots)
 V_p (maneuvering) 112 m.p.h. (97 knots)
 V_{fe} (flaps extended) 95 m.p.h. (82 knots)

C. G. range Normal: (+17.5) to (+23.0) at 2000 lb.
 (+12.0) to (+23.0) at 1800 lb.
 (+9.5) to (+23.0) at 1400 lb. or less

GROSS WEIGHT
 (lb.)



EMPTY WEIGHT Normal: 1136 lb.
 MAXIMUM WEIGHT Normal: 2000 lb.
 NO. SEATS 4 (2 at +21, and 2 at +49).
 MAXIMUM BAGGAGE 100 lb. (+67)
 FUEL CAPACITY 36 gal. (2 Wing tanks at +24) (216 lb)
 OIL CAPACITY 2 gal. (-29) (15 lb)
 CONTROL SURFACE MOVEMENTS Stabilizer 1° Up 6 1/2° Down
 Elevator 24° Up 12° Down
 Aileron 15° Up 15° Down
 Rudder 16° Right 16° Left
 Flap 40° Down

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See CARS sections for proper inspection schedules..

<https://tc.canada.ca/en/corporate-services/acts-regulations/list-regulations/canadian-aviation-regulations-sor-96-433/standards/standard-625-appendix-b-maintenance-schedules-canadian-aviation-regulations-cars>

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